

23 Abstract

24 Ammonia (NH_3) is regarded as an alternative fuel not only as a carbon-free fuel but also as a
25 renewable hydrogen-carrier. It is possible that the safety in micro-combustor can be improved
26 through partial NH_3 substitution for hydrogen. However, knowledge of the thermal
27 performance and nitrogen oxides (NO_x) emission of ammonia/hydrogen combustion,
28 especially in the micro-combustor, has been insufficient. In order to enhance thermal
29 performance, reduce NO_x emission and improve flame stabilization of ammonia/hydrogen
30 fuelled micro-combustors for thermophotovoltaic (TPV) application, three types of micro-
31 combustors with a wavy profile are designed and evaluated. For this, a three-dimensional (3D)
32 numerical model with a detailed chemical reaction mechanism has been verified and applied
33 to assess the thermal performance of the modified micro combustors in terms of the outer
34 wall temperature distributions. The average temperature of these wavy combustors is found
35 to be much higher than that of the conventional smooth combustor, regardless of the
36 hydrogen/ammonia mixture flow velocity. Moreover, the wavy is a more effective measure to
37 improve temperature uniformity when the mixture velocity is greater than 12m/s. Comparing
38 the flame stability behaviours of hydrogen/ammonia/air blended combustion in both the
39 conventional and the proposed wavy combustors reveals that the blowout limit is effectively
40 broadened. Finally, the effects of 1) hydrogen/ammonia blended ratio and 2) fuel-air
41 equivalence ratio on NO_x emissions are examined in detail. It is found that approximately
42 21.2% of NO_x emission reduction could be achieved in the ARC wavy micro-combustor.
43 NO_x emission reduction can be gradually improved, as the nitrogen fuel mass ratio is
44 increased. This present research sheds lights on an effective design of a micro-combustor
45 with enhanced thermal performance and reduced NO_x emission.

46 **Key words:** *energy conversion performance; blended fuel; micro-combustion; wavy profile;*
47 *blowout limit; NO_x emissions.*

48 **1. Introduction**

49 Ammonia (NH₃) is regarded as an alternative fuel not only as a carbon-free fuel
50 but also as a renewable hydrogen-carrier. NH₃ is also considered as a clean and
51 high density hydrogen carrier since it can be easily rendered liquid at
52 atmospheric temperature and because of its lower cost of storage and production,
53 distribution and transportation [1]. Besides, NH₃ is able to burn directly with
54 blends fuel or additives [2]. It is possible that the safety of hydrogen (H₂) use in
55 combustor can be improved through partial NH₃ substitution for hydrogen. With
56 the increasing demand of worldwide industries for portable energy conversion
57 and power-harnessing in smaller-scale applications, micro power generators
58 (MPG) are developed using micro or nanotechnologies to convert different
59 forms of energy into electricity energy [3]. In recent years, it is essential to
60 establish efficient approaches for designing and manufacturing energy-efficient
61 and environmentally-friendly solutions for MPG. The micro-
62 thermophotovoltaic (MTPV) [4, 5] is one kind of micro-combustion-based
63 MPG attracting extensive attention due to its reliability, longevity and simple
64 structure without moving parts, and is suitable for use in commercial electronics
65 and personal micro devices [6]. Researcher has experimentally evaluated the
66 feasibility of using ammonia-hydrogen blend-fuelled as green fuel in MTPV

67 device with GaSb, modifying a previous hydrocarbon fuel combustor for MTPV
68 [7].

69 Combustion of hydrogen or mixture fuels provides a much higher energy
70 density due to the significant reduction in weight or volume of combustor
71 chamber compared with portable lithium-ion batteries [8]. Temperature
72 distribution characteristics are key points to improve the combustor efficiency
73 [9] and the performance of the MTPV. The major challenges faced in
74 optimizing the micro-combustor's ability to convert chemical fuels into
75 electricity energy are not only to achieve uniform and high temperature
76 distribution [10] and wider flammability limit [11] but also to obtain higher
77 emission efficiency of exhaust gas [12].

78 One of the most important issues for optimization of combustor is to improve its
79 outer wall thermal performance, which attracts considerable experimental and
80 numerical investigations due to its direct effect on feasibility and demanded
81 energy conversion efficiency. Various new methods have been applied to
82 combustors by employing fins or bluff body [13, 14], catalytic combustion [15],
83 porous media [16-18], counter-flow [19], and heat recirculation [20, 21]. Peng
84 et al. [16] inserted a porous media into a combustor to augment the wall
85 temperature and subsequently enhanced the applied radiation energy of MTPV.
86 This is attributed to the solid matrix of porous media offering a better thermal
87 characteristic (heat capacity, conductivity, and emissivity) than gas. Yan [22]

88 and Yilmaz [12] reported that adjusting the angle and width of the bluff body
89 can improve the combustion efficiency and blowout effectively. Li and Hong
90 [23] conducted numerical study of H₂/CH₄/air blended combustion and
91 concluded that the output electrical energy of the combustor with the
92 implementation of catalyst (Pt) is increased by up to 23.8% compared with that
93 without the catalyst. They also revealed that the catalyst has adverse effects on
94 combustion process at higher mass flow rates. The objective of Akhtar's [24]
95 research was to investigate the impact of different cross-section types
96 (including circular, rectangular, square, trapezoidal and triangular) on the outer
97 wall temperature distribution and the flame structure of hydrogen fuelled micro-
98 combustor. The results showed that trapezoidal and the triangular cross-sections
99 brings about better thermal performance and overall energy conversion
100 efficiency for the low flow rates and high flow rates respectively. Zuo *et al.* [25]
101 studied the positive effects of H₂/air premixed flame in the combustor improved
102 by gradually varying the diameter. Mansouri *et al.* [26] proposed configuration
103 with wavy profiles as thermal performance enhancement for flat micro-
104 combustor. The significant impact brought by the above-mentioned
105 optimization of inner combustor on the thermal performance was observed. The
106 lack of an easy implement approach to improve the thermal performance of
107 micro combustors partially motivated the present work.

108 Blowout limits [8, 27] of micro-combustors need to be considered for feasible
109 industry operation by researchers and engineers. The residence time of gas
110 mixture decreases as combustor size decreases. Furthermore, excessive heat loss
111 from the outer wall of combustor will suppress flame stability and even result in
112 blow out [28]. Massive research has been conducted to extend the flame
113 stability of the conventional micro-combustors [8, 27, 29]. Therefore, the
114 current topic of interest lies in approaches that are able to effectively broaden
115 blowout limit and enhance thermal performance.

116 In several previous studies [10, 30-35], ammonia/hydrogen was seen as one of
117 the promise fuel and it was actually used for combustor design. NO_x emissions
118 produced via combustion present significant risks to both health and the
119 environment. Therefore, although ammonia is an important enhancer of
120 combustion, there is a large NO_x proportion of emissions using this fuel [36].
121 **Cai et al.** [37] conducted numerical investigation on the performances and NO
122 emission characteristics of ammonia/oxygen premixed combustion. **Xiao et al.**
123 [38] and **Rocha and Ramos** [39, 40] reported a study of detailed mechanisms of
124 ammonia/air to represent the reaction kinetics and NO emission under
125 combustor conditions. **Li et al.** [41] found that the NO_x emission increases
126 sharply with the NH₃ addition mainly through the enhanced HNO pathway in
127 the main reaction zone. **Hayakawa et al.** [42] explored the knowledge of
128 characteristics of ammonia/air flames at the high pressures. **Valera-Medina et al.**

129 [43] performed on practical swirl combustors with ammonia fuel mixture to
130 optimise the combustion stability and emission performance.

131 The major scopes of this paper are not only to choose the best possible
132 configuration with respect to energy efficiency, but also to obtain optimum fuel
133 condition with consideration of lower NO_x emissions. It is clear that further
134 efforts are still needed in the NO_x concentration reduction in order to avoid an
135 increase in NO. Recently, with the advances in the Additive Manufacturing (3D
136 printing) technologies for micro parts, research efforts are moving toward the
137 optimization of micro combustor by easy and reliable methods. The optimal
138 design is obtained for the first step. In this paper, the micro-combustors with
139 three different wavy profiles are proposed firstly to improve performance. Then
140 the average temperature distribution and uniformity over the outer wall of
141 hydrogen-fuelled micro combustor with different wavy types are compared for
142 various inlet flow velocities. The detailed comparison analysis of the flame
143 structure and temperature distributions in these combustors is carried out to
144 explain the reason of enhancement brought by the improved wavy combustors
145 in comparison with the conventional cylinder combustor (REF). To validate the
146 effect on flame stability of the improved wavy micro-combustor, the blow-out
147 limit is applied to compare and investigate the combustion characteristics of
148 ammonia/hydrogen/air blended between the REF and newly proposed ARC
149 combustors. Finally, the evaluation of reduced NO_x emissions efficiency for

150 wavy micro combustor is performed. The effects of fuel parameters including
151 blended ratio and equivalence ratio on NO_x emissions are also analysed in detail.

152 2. Methods and Model

153 2.1 Geometric model

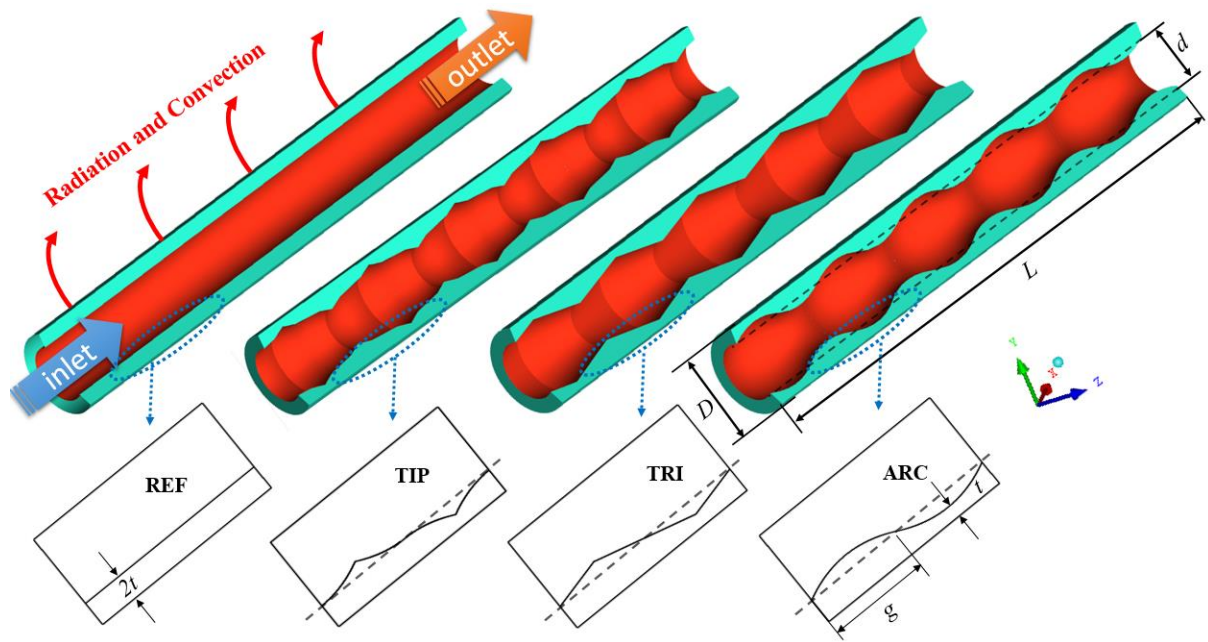


Fig. 1 Schematic diagram of four micro-combustors and partial cross-section view of the four wavy profiles.

154 The reference micro-combustor is an axisymmetric cylinder for MPG in
155 industrial application. The fuel flows into from the inlet and combustion in the
156 inner circular channel part. Then the energy of radiation and convection is
157 released from the outer wall. The schematic diagram of the micro-combustor
158 without or with the inner wavy profile and their various shapes are presented in
159 Fig. 1. Three improved micro-combustors are designed to keep periodical inner
160 wavy profile. The inner wall of improved combustor has five continuous waves

161 equally spaced in the axial direction. All four combustors with the same fluid
 162 region volume are chosen for the present research. The cross-sectional views of
 163 all four combustors are shown in Fig. 1. The conversional reference combustor
 164 is named REF. The TIP shape is connected by arcs to form a tip, and the TRI
 165 shape consists of linear edges of a triangle. The profile of the inner surface of
 166 ARC is connected by smooth arcs to facilitate wavy smoothness. During the
 167 optimization design of the combustor, the volume of the entire internal channel
 168 and wall material are respectively kept constant. The wavy profile is used to
 169 change the contact surface between the flame and combustor. The length of the
 170 micro-combustor is L , the external diameter is D , the internal diameter is d , the
 171 minimum wall thickness is t , and the periodical wavy length is $2g$. The micro-
 172 combustors are made of 316 stainless steel whose properties are given in the
 173 reference [14]. Table 1 shows detailed dimensions of the combustors.

174

Table 1. Dimensions of combustor.

Dimensions	Values (mm)
D	3
d	2
L	20
g	2
t	0.25
<i>wavy periods</i>	5

2.2 Mathematical Model

A detailed chemical mechanism and transport based 3D CFD model is adopted in the present numerical investigation. Considerable efforts have been spent to explore thermal performance and reduce NO_x emissions of different optimized micro-combustors. All the issues below are mandatory to ensure the accuracy and reliability of the numerical results, e.g. the numerical method, turbulence models, mechanism and boundary conditions, which need to be considered properly. The following issues are considered and assumed before establishing the numerical model [44]: (1) fuel/air mixture steady combustion; (2) considering outer wall radiation and convection; (3) incompressible flows; (4) neglecting surface reaction effect. The fuel inlet Reynolds number can be estimated as Kuo and Ronney [45] modelled for Swiss-roll micro combustors. It is observed that modelling of turbulent flow and transport at $Re > 500$ is necessary to keep consistency between the experimental and numerical results. Thus the standard $k-\varepsilon$ model [31] is used for the turbulent flow and heat transfer in the present study. With the above assumptions, the mathematical model of hydrogen/ammonia/air premixed combustion in micro-combustor is established by using the mass, momentum, energy species and equilibrium law. The related governing equations are written as follows [26].

Mass:

$$\nabla \cdot (\rho \vec{v}) = 0 \quad (1)$$

195 where ρ is the gas density and \vec{v} is the velocity in the v direction.

196 Momentum:

$$\rho(\vec{v} \cdot \nabla \vec{v}) = -\nabla p + \nabla \cdot \left(\mu \left[\nabla \vec{v} + (\nabla \vec{v})^T - \frac{2}{3} \nabla \cdot \vec{v} \mathbf{I} \right] \right) \quad (2)$$

197 where p is the gas pressure, μ is the viscosity and \mathbf{I} is the unit tensor.

198 Energy species:

$$\nabla \cdot \vec{v} (\rho E_f + p) = \nabla \cdot \left[\lambda_{\text{eff}} \nabla T - \left(\sum_j h_j \vec{D}_j \right) + \left(\mu \left[\nabla \vec{v} + (\nabla \vec{v})^T - \frac{2}{3} \nabla \cdot \vec{u} \mathbf{I} \right] \right) \cdot \vec{v} \right] + S_f^h \quad (3)$$

199 where E_f is the total energy, λ_{eff} is the effective thermal conductivity of fluid,

200 ∇T is the temperature difference between fluid and walls. h_j is the enthalpy of

201 species j , \vec{D}_j is the mass diffusion coefficient of species j , \vec{u} is the velocity in

202 the u direction, the and S_f^h is the source term of the fluid enthalpy.

203 Wall energy equation:

$$\nabla \cdot (\lambda_w \cdot \nabla T) = 0 \quad (4)$$

204 where λ_w denotes the conductivity of the combustor surface.

205 Species equation:

$$\nabla (\rho \vec{u} Y_j) = -\nabla \vec{D}_j + \omega_j \quad (5)$$

206 where ω_j and Y_j are the net production rate and the mass fraction of species j .

207 Ideal gas equation:

$$p = \rho R_g T \quad (6)$$

$$R_g = \frac{R_0}{M} \quad (7)$$

208 Where R_g denotes **the universal** gas constant, M is **the** molecular mass.

209 The outer wall average temperature:

$$T_{ave} = \frac{\sum A_i T_i}{\sum A_i} \quad (8)$$

210 The outer wall temperature uniformity is evaluated by the standard deviation
211 parameter of wall temperature σ_T , which can be written as:

$$\sigma_T = \sqrt{\frac{\sum_{i=1}^N A_i (T_i - T_{ave})^2}{\sum_{i=1}^N A_i}} \quad (9)$$

212 where A_i and T_i are the area and temperature of outer wall cell i respectively,
213 and T_{ave} is the average temperature of outer wall.

$$Q = h_0 A (T_w - T_0) + \varepsilon \sigma A (T_w^4 - T_0^4) \quad (10)$$

214 Where h_0 is the natural convection coefficient, σ denotes the Stephan-
215 Boltzmann constant which equals to $5.67 \times 10^{-8} \text{ W / m}^2 \cdot \text{K}^4$, ε denotes the emissivity
216 of the combustor wall, T_0 and T_w **are** the ambient temperature and outer wall
217 temperature respectively.

218 A specie production expression model for the net rate has to be used to close Eq.
219 (4). The turbulence dissipation e and turbulent kinetic energy k for k-epsilon
220 model are obtained by solving transport equations. The pressure and velocity

221 coupling are calculated by using SIMPLE algorithm handled by Fluent (Ansys
222 19.0). These governing equations are solved by the finite volume and the
223 implicit solution method of under-relaxation. A second-order-upwind scheme is
224 used to discretize all governing equations together with the essential relations
225 and proper boundary conditions. Konnov [46] focused on the more detailed
226 mechanism of NH_3 and its product. Kumar and Meyer [47] used the Konnov
227 mechanisms for $\text{H}_2/\text{NH}_3/\text{air}$ mixtures over a wide range equivalence ratios to
228 compare the experimental data and models and evaluate its accuracy. A detailed
229 mechanism model [48] involving 38 species and 232 reactions for
230 hydrogen/ammonia/air gas-phase chemical reaction is verified and achieved by
231 Nakamura and Shindo [49]. It is also the fundamental mechanism model to be
232 applied and verified in the present simulation. The specific heat capacity and
233 mass diffusivity of fuel and air mixture are calculated by the mixing-law and the
234 kinetic-theory respectively. The energy balance between the atmosphere and
235 combustor will be achieved once the flame is stabilized in the micro-combustor.
236 Based on the suggestions from Bongers and De Goey [50] on micro-combustion,
237 Dufour effect is ignored, while Soret effect is considered since it has an impact
238 on NO production rate. The body force and heat transport caused by the
239 concentration gradient are neglected due to their small amount as well as the
240 pressure diffusion[51]. The mass fraction-weighted average of all species is
241 selected for calculations of the mixture gas thermal conductivity and viscosity

242 [52]. The energy equation convergence criterion is set as 10^{-6} , the other
243 continuity, specie and turbulence equations criterion is 10^{-5} .

244 **2.3 Computation Domain and Boundary Conditions**

245 The physical computation model contains a fluid domain and a solid domain.
246 For the fluid domain, heat and mass transport of fuel/air mixture along with the
247 chemical reactions are considered. The solid domain takes into account the heat
248 transport encompassing conduction, convection, and radiation. A representative
249 ARC computational domain of the physical system analysed in this study is
250 presented, while other configurations considered in this study such as REF, TIP
251 and TRI are not shown. The mesh schemes of both domains are shown in Fig. 2,
252 in which non-uniform structured mesh is outputted by the pre-processor
253 meshing program ICEM. To obtain a geometric model of the combustor as
254 accurately as possible, various blocks are divided to be close enough to
255 represent the geometric characteristics. These blocks also facilitate the
256 controlling of the number and scale of cells. All correlation vertexes and edges
257 are stitched together into the block of combustor. The cells in all blocks are
258 meshed with high-orthogonality hexahedron, with refinement in the parts of
259 high strains or variations so that the numerical errors can be minimized. Solid
260 domain adopts medium density while fluid domain adopts uses a fine density,
261 because the flame zone within the fluid domain is more complicated than that in
262 the solid domain. **In the streamwise direction**, a higher mesh distribution is used.

263 Also, it can be seen that the mesh sizes of the interfaces between the solid
264 domain and fluid domain are nearly the same, which is helpful for the heat
265 transfer couple between the interface of the fluid domain and the solid domain.

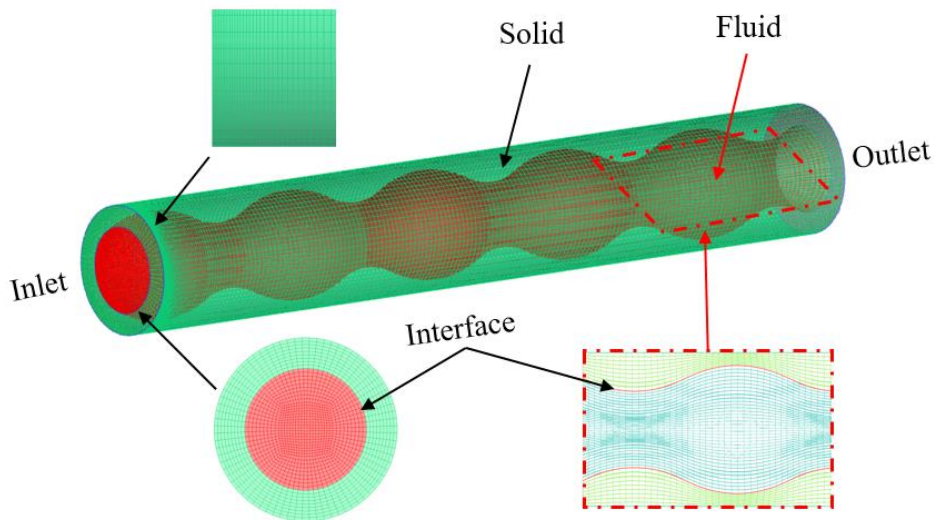


Fig. 2 Geometric domain scheme model and mesh details

266 From the enlarged region, it is observed that the region near the inlet of both
267 fluid and solid domains has finer mesh than that of other regions. This is
268 because that the gases inject condition from inlet is a significant initial value in
269 the numerical results. The inner surface of solid region and outer fluid domain
270 surfaces are set as interface to interpolate values. The main boundary conditions
271 of all combustor are taken as the velocity-inlet and pressure-outlet respectively.
272 The inlet velocity is variable to evaluate its effect on the combustion
273 performance. The pressure and O_2 mass fraction of outlet are set according to
274 the air condition in atmosphere. All detailed boundary condition settings are
275 generalized in Table 2.

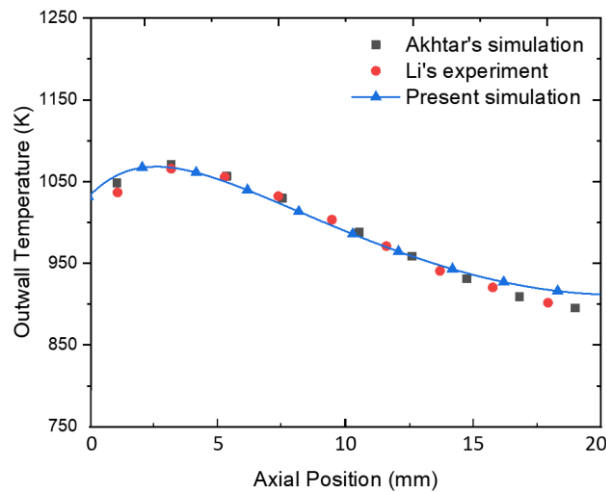
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Table 2 Boundary conditions.

Types	Parameters	Values
	Mass fraction of each species	regard to H ₂ or H ₂ /NH ₃ air equivalence ratio(0.6-1.2)
Inlet	Velocity (m/s)	regard to cases (8-20)
	Temperature (K)	298
	Turbulent intensity (%)	5
	Hydraulic diameter (mm)	2
	Gage pressure (Pa)	101325
Outlet	Turbulent intensity (%)	5
	Hydraulic diameter (mm)	2
Inner wall	Wall slip	Non-slip
	Thermal condition	Coupled
	Material	316 Steel
Outer wall	Thermal condition	Mixed
	Heat transfer coefficient (W/(m ² ·K))	20

277

2.4 Model Validation and Mesh Independence Study



278

279 **Fig. 3 Comparison of the outer wall temperature between the present simulation and literature**
280 **data.**

281 The results of the present numerical simulation are compared against the
282 corresponding experimental and numerical data which comes from Li *et al.* [53]
283 and Akhtar *et al.* [24] to verify the accuracy of the computational model. The
284 length of the main part combustor from the experimental setup is 20 mm, and a
285 7 mm inlet is also arranged. The main part combustor size and wall thickness
286 are both the same as the present combustor. In the experimental study of Li *et al.*
287 [53], temperature distribution of experimental data on outer wall is showed.
288 Akhtar *et al.* [24] conducted a 2-D axisymmetric numerical simulation on the
289 same micro-combustor too. In this paper, **the pure hydrogen case with an inlet**
290 **velocity of 12m/s and hydrogen/air equivalence ratio of 0.8 is selected.** Fig. 3
291 presents the temperature distribution profiles of the external wall, showing both
292 the experimental and simulation data **along the axial direction.** It can be seen
293 that the consistence between the present simulation and experimental results

294 available in the literature is excellent, showing the similar trend of temperature
 295 distribution and the position of the temperature peak value. **The maximum**
 296 **temperature difference occurs near the combustor outlet, with a value of 19 K,**
 297 **i.e.,** the maximum relative error between the numerical and experimental results
 298 is only 2.1%, which indicates that the computational model established in **the**
 299 present study is reliable.

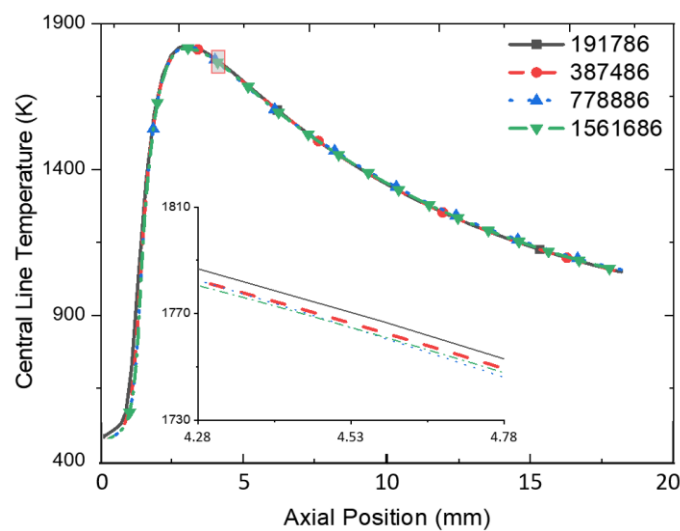


Fig. 4 Distribution of the central line temperature with different grid sizes under the inlet velocity of 12m/s.

300 **To exclude the influence of the grid size on the calculated results,** a mesh
 301 independence test with 4 different mesh densities ranging from 191,786 to
 302 10,561,686 is conducted. Fig. 4 shows the effect of mesh scale on the central
 303 line temperature of the micro combustor where **the mixture inlet velocity** and
 304 the hydrogen/air equivalence ratio are kept at 12m/s and 0.8, respectively. A
 305 sizable variation in the central line temperature at streamwise direction for
 306 different mesh sizes is observed in the vicinity of the flame, i.e., the flame

307 distance. The upstream of the flow is also relatively mesh sensitive. By
308 checking the centreline gas temperature profiles of the above meshes (zoomed-
309 in in Fig. 4), it is found that the temperature is a bit decreased with the increase
310 of grid size. It is observed that the mesh with about 387,486 cells is able to well
311 illustrate the gas mixture combustion temperature profiles, and a higher number
312 of cells (778,886) does not yield any significant improvement. It was concluded
313 that the mesh size level of 387,486 cells should be considered suitable for all the
314 following numerical computation sections since it provides a perfect trade-off
315 between the computational accuracy and the time cost. Furthermore, the
316 corresponding mesh size is selected in the present geometrical models with
317 different wavy profiles.

318 **3. Results and Discussion**

319 **3.1 Wavy Profile Effect on Thermal Performance**

320 The output of applied energy from micro TPV is proportional to the thermal
321 performance from the outer wall of combustor. The mean wall temperature is a
322 significant symbol of thermal performance, which is defined as the mean value
323 of all cells of the combustor wall. Fig. 5 presents the outer wall mean
324 temperature values of the combustors employing varied wavy profile
325 configurations. In this part, the inlet velocity varies from 8 m/s to 20 m/s, and
326 H_2 /air equivalence ratio is set as 0.8. As shown in Fig. 5, **with increasing the**
327 **inlet velocity, the average temperature values increases significantly**, whether

328 the micro-combustor has any wavy or not. Under the same equivalence ratio in
329 this section, the increased inlet velocity results in a higher wall temperature.
330 This is because the increasing of the inlet velocity provides a higher energy
331 input, and thus release more heat generation from the combustion. It is clear that
332 the change of wavy type also affects the temperature distribution. It also
333 demonstrates that regardless of the fuel flow velocity, the average temperatures
334 on the wall of TIP, TRI and ARC combustors are higher than that of REF
335 combustor.

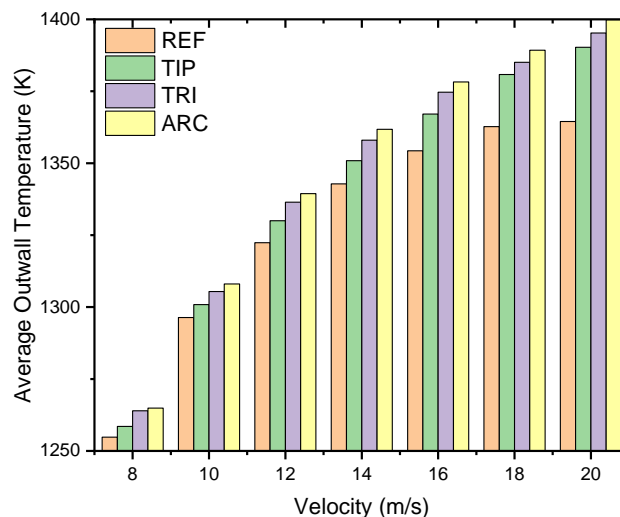


Fig. 5 Average outer wall temperature profiles for different wavy combustors.

336 The combustor with ARC wavy obtains the highest temperature, while the value
337 of REF combustor is the lowest. The average temperature in REF is increased
338 from 1255K to 1364K as the velocity increases from 8 m/s to 20 m/s. On the
339 contrary, the ARC combustor reaches 1400K at the velocity of 20 m/s. A
340 maximum improvement of 36K can be achieved when comparing the ARC

341 wavy combustor with the conventional combustor. That is, the improvement of
 342 output performance on ARC-TPV is achieved as 10.98%. The improvement of
 343 average temperature of wavy is more obvious with the increase of mixture
 344 velocity. It should be pointed out that wavy profile enlarges interface between
 345 fluid and solid zone for heat transfer. In addition, the average temperature
 346 improvement is due to the formation of a recirculation zone on the concave
 347 wavy wall which provides more residence time and heating time to stabilize and
 348 enhance heat transfer. The viscosity caused turbulence enhances heat transfer
 349 between the combustion gas and the solid wall. The thermal performances of
 350 the micro-combustors are not only sensitive to the mixture velocity, but also to
 351 the change of interior structure area in the combustion chamber.

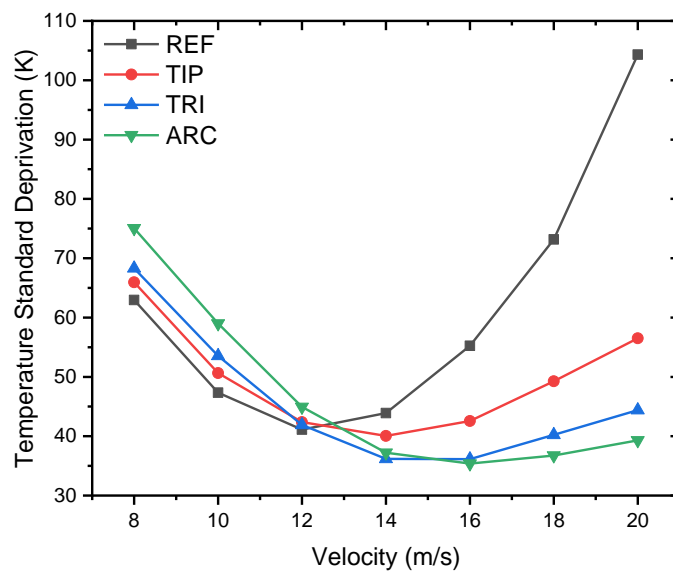
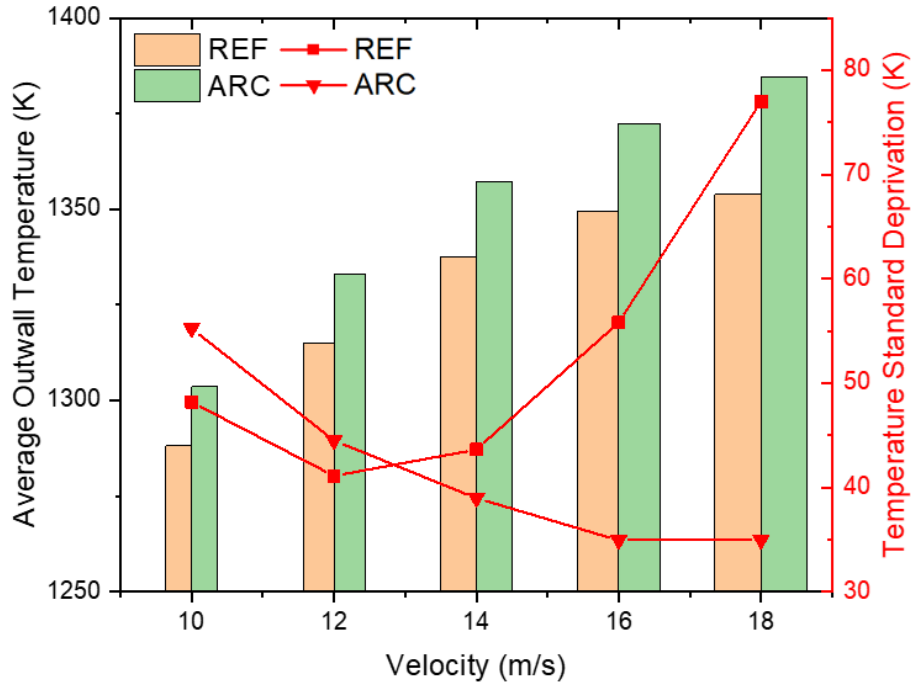


Fig. 6 Non-uniformity of outer wall temperature distribution.

352 The higher and more uniform outer wall temperature has an imperative effect on
353 the radiation efficiency and output power of the micro-TPV system [33]. In
354 order to achieve a high energy conversion efficiency, the wall temperature
355 uniformity must be improved. Furthermore, a uniform temperature distribution
356 reduces outer wall thermal stresses of the micro combustor and prolongs the
357 operation period [54]. In this paper, the temperature standard deviation is
358 measured by calculating uniformity coefficient to quantify the non-uniformity.
359 For this reason, the variation of wall temperature distribution non-uniformity
360 under various inlet velocities is presented in Fig. 6. With an increase of the inlet
361 velocity, the temperature standard deviation (the lower derivation indicates
362 better uniformity) decreases first in the velocity range of 8-12m/s, then
363 increases in the velocity range of 13-20m/s. It can also be observed that
364 temperature standard deviation is very high for REF micro combustor without
365 any wavy. It can be easily observed that the highest standard deviation of about
366 105 K is achieved at the velocity of 20m/s for reference REF combustor. In
367 addition, the non-uniformity coefficient drops down from 42K to 34K due to the
368 implementation of the ARC wavy. Wavy profile is a highly effective measure to
369 improve uniformity when the mixture inlet velocity is higher than 12m/s. The
370 effect of the inner wall on thermal performance is related to the fuel inlet
371 velocity. Although the wavy profile has a negative effect on uniformity at a low
372 speed on, it is obvious that the positive effect is generally much more obvious
373 than the negative effect. This can provide a reference for the further design of

374 micro-combustor. For temperature uniformity, it is mostly likely due to the
375 increase of local turbulence gradient. The turbulent intensity distribution and for
376 hydrogen-fuelled combustion in REF and ARC combustor are compared. It can
377 be seen that at a lower velocity (8m/s), wavy just causes increasing of
378 turbulence intensity at local bulges. The local turbulence gradient leads to a
379 negative effect on uniformity under lower-inlet-velocity condition. At a higher
380 velocity (20m/s), the turbulence intensity is enforced over a wide area rather
381 limited in local area. It is attributed to the fact that the positive effect of wavy
382 profile is sizable enough to change the fluid movement towards the outlet and
383 enhance the heat transfer between the combustion chamber wall and the
384 combustion gas.

385 Fig. 7 shows the outer wall temperature distribution and non-uniformity for
386 20%NH₃+80%H₂ blended fuelled combustion. These trends are similar to the
387 above-mentioned pure hydrogen combustion. According to the foregoing
388 analysis, ARC is a better configuration for either hydrogen/air or
389 ammonia/hydrogen/air fuelled combustion considering temperature uniformity
390 and average temperature.



391

392 **Fig. 7 Outer wall temperature distribution and non-uniformity for 20%NH₃+80%H₂ fuelled.**

393 **3.2 Thermodynamics characteristics**

394 To explore the detailed effects of wavy profile on both wall temperature
 395 distribution and flame temperature, contours of flame temperature field
 396 variation of middle cross section on four combustors are shown in Fig.8, where
 397 fuel flow velocity and equivalence ratio are set as 20m/s and 0.8, respectively.
 398 High temperature values associated with wavy wall are mainly rooted in the
 399 above mentioned thermal performance. Convergent-divergent flow channel with
 400 wavy profile affects inner reactants velocity, provides more residence time and
 401 heating time, and makes a fundamental change in the interaction amplitude
 402 between the wall and the shape of the flame evolution. To further analyze the
 403 basic mechanism that leads to the difference in temperature distribution of the
 404 four kinds of micro-combustors, several radial cross-section temperature

405 distributions under different combustor structures are revealed. The wall
 406 temperature at $X=2$ cross section of the REF micro-combustor is lower than
 407 that of the other three combustors. The wall temperature distribution at $X=8$ and
 408 10 of the TRI and ARC micro-combustor is more homogeneous. The reason for
 409 these abovementioned phenomena is that the flame radial is narrowed by the
 410 internal wavy profile. The flame front is compressed shorter and the ignition
 411 zone (green zone in the contour) is broadened.

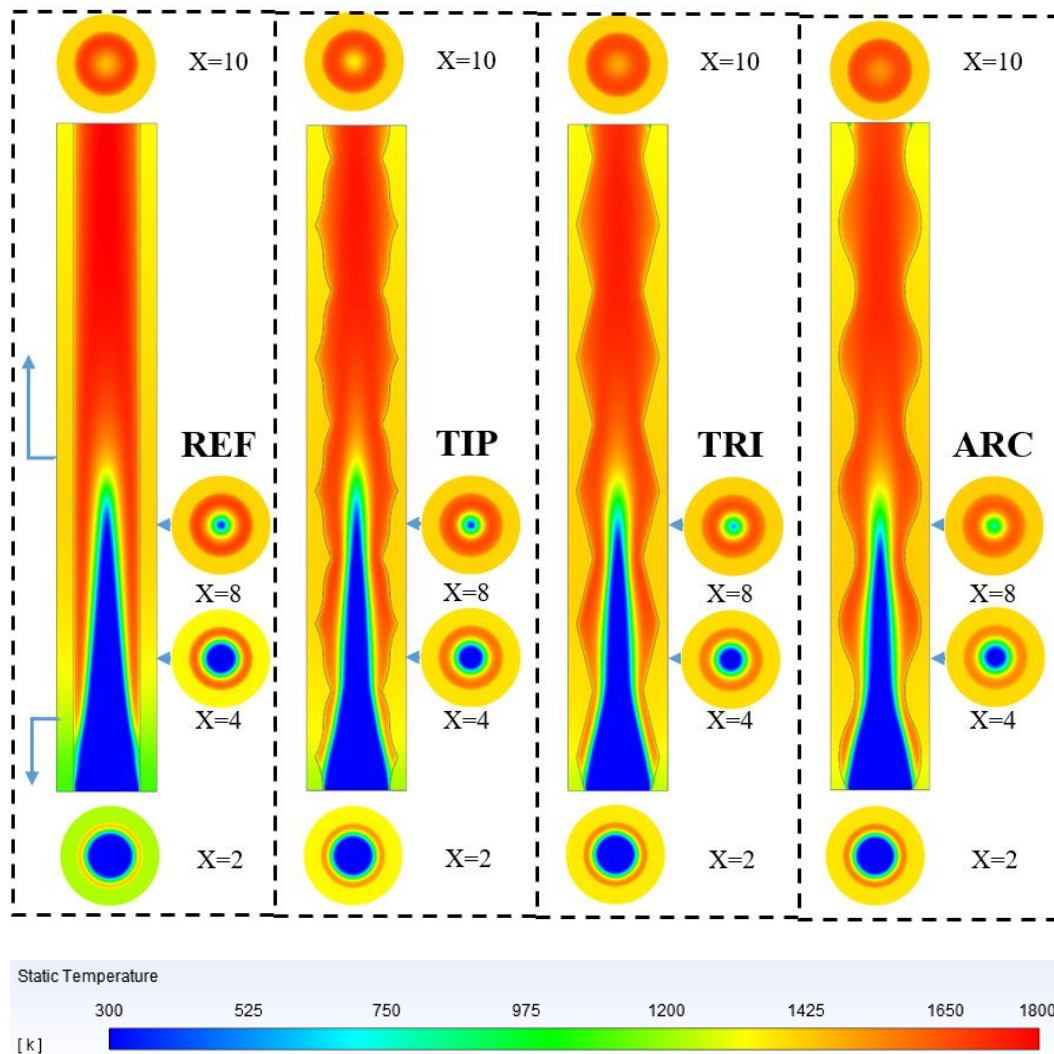


Fig. 8 Variation of the temperature field.

412 It is **worthwhile to mention** that under the same fuel flow rate, the increase in
413 combustion gas velocity due to the periodical change of inlet area does result in
414 **a** higher temperature of outer wall near the inlet compared to REF combustor.
415 This phenomenon shows that the wavy arrangement is beneficial to the
416 preheating of the mixture by the higher temperature inlet wall. **In general**, the
417 variation resulted in a more intensive heat transfer between the solid wall and
418 combustion gas.

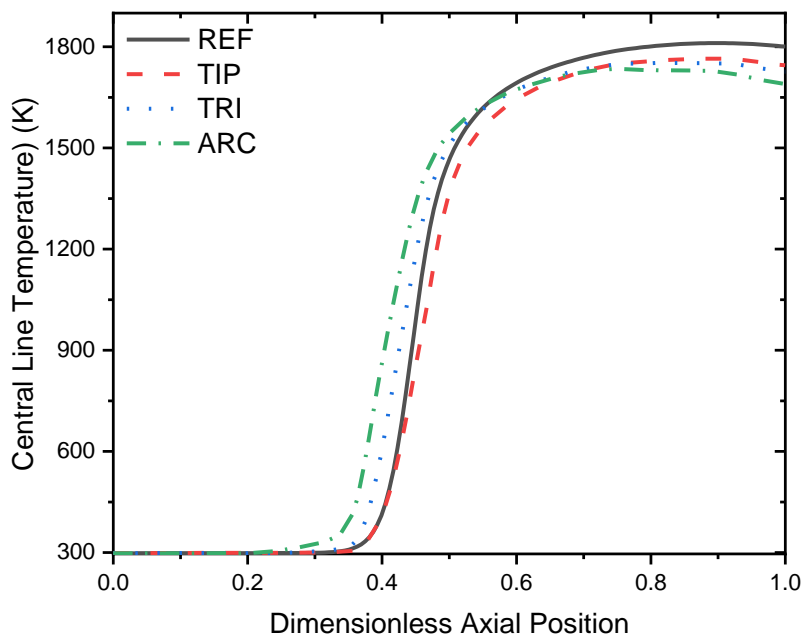


Fig. 9 Temperature profiles along the central line of different wavy profile shapes.

419 Fig. 9 compares the temperature profiles along the central line in the four above
420 mentioned combustors. The flame reaction zones in TRI and ARC are almost
421 located in the upstream region along the central line due to the function of two
422 wavy profiles. However, the flame front edge moves downstream with the

423 implementation of TIP wavy profile. The flame location is sensitive to gas
424 velocity. After checking the velocity distribution of the combustor cross section,
425 it can be seen that the wavy wall has an obvious influence on the flow velocity.
426 The front edge of the flame is located in the middle of the combustor, which is
427 exactly in the accelerating area of the gas from the contraction passage. It can be
428 concluded that the acceleration area at the TIP's central passage is closer to the
429 downstream. Thus the flame front edge region moves downstream with the
430 implementation of the TIP wavy profile comparing with TRI and ARC.
431 Meanwhile, it can be observed that the highest flame temperature decreases
432 with the function of wavy profile because more area is used to transfer heat to
433 the outer solid walls. It can also be seen that the change of the inner temperature
434 is consistent with the variation of the temperature field in Fig. 8.

435 3.3 Flammability Behaviour

436 Fig. 10 presents the combustion states of REF and ARC micro-combustors with
437 different hydrogen/ammonia mass blend ratios and velocities. Combustion
438 states are similar in cases of 100% hydrogen and 80% H_2 +20% NH_3 . REF and
439 ARC have the same flammability. As the blend proportion of H_2 decreases to
440 lower than 80%, the combustion states in REF begin to blow out. Moreover, the
441 value of blow out limit becomes smaller with the decrease of H_2 . The blow out
442 limits of REF micro-combustor operation on 60% H_2 +40% NH_3 and
443 40% H_2 +60% NH_3 are 10 m/s and 16m/s, respectively. This indicates that

444 ammonia has a lower flame velocity than hydrogen. Beyond the blow out limit,
 445 the flames attached to the inner combustor fail to keep stable and contracts
 446 down toward complete extinction. Additionally, by comparing the combustion
 447 states of ARC combustor, the most obvious difference is revealed: ARC can
 448 **combust** stably in any fuel mixture ratio or velocity due to the varied velocities
 449 and longer residence time in channel. This is **attributed** to the preheat-process of
 450 wall on the hydrogen-fuel, and turbulence enhances the flame stability. The
 451 present simulation results shows that the hydrogen can **combust** with inlet
 452 velocity of 20m/s. If the ammonia ratio in the mixture-fuel rises to a certain
 453 level, the flame even will be blown out at 12m/s. Therefore, **a** micro-combustor
 454 with ARC is more favourable to expand the blow out limit and anchor flame
 455 than the conventional REF combustor.

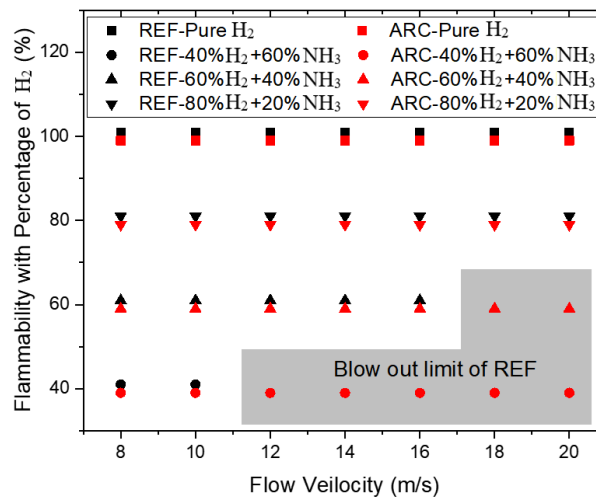


Fig. 10 Blowout limit of REF and ARC with different hydrogen/ammonia mixture and velocities.

3.4 NO_x Emissions Reduction

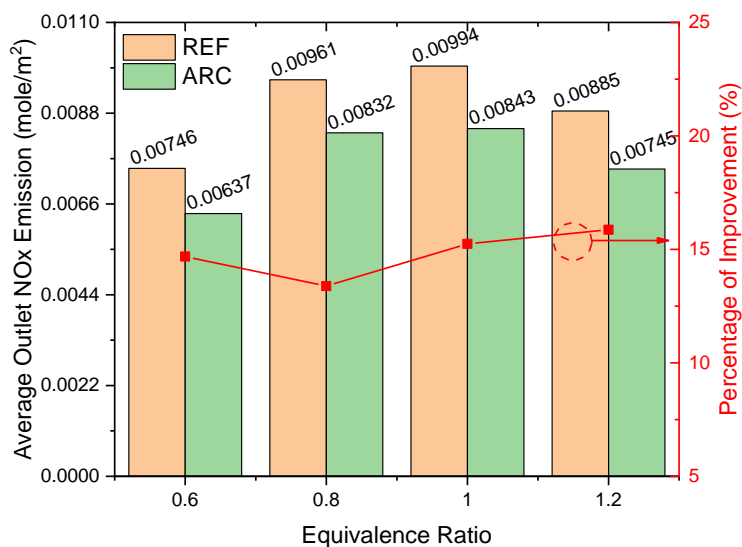


Fig. 11 Variation of NO_x emissions under different equivalence ratios of H₂/NH₃.

457 For combustion with enhanced performance, NO_x control and emission
 458 reduction technology is also a relatively complex issue. Fig. 11 attempts to
 459 verify the role of wavy structure on NO_x pollution prevention of the exhaust gas
 460 by using 60%H₂+40%NH₃ mixture fuel in REF and ARC combustors. Both
 461 REF and ARC combustors have the same area of outlet. Effects of combustor
 462 with wavy or without, and equivalence ratio **are** examined in this section. The
 463 nitrogen oxide analysis of the outlet exhaust gas shows that the content of NO₂
 464 is almost zero emission, but there is more NO. The NO is represented as the
 465 NO_x emission for following analysis. In the field of micro combustor, the off-
 466 stoichiometric condition is intentionally used to protect the combustor wall from
 467 hot combustion gases. Equivalence ratio of fuel flow for combustion has been
 468 correlated to the amount of NO_x generated. To understand the NO_x emission

469 characteristics for various conditions, the global equivalence ratio is varied in
 470 the range of 0.6~1.2 together with the change of combustor configuration. With
 471 equivalence ratio increasing from 0.6 to 1.2, the REF combustor outlet NO_x
 472 emission increases from 0.00746 mol/m² to 0.00994mol/m², then decreases to
 473 0.00885 mol/m², the peak emission value is at stoichiometric equivalence ratio
 474 of 1.0. Regardless of the fuel equivalence ratio, applying the ARC wavy to fuel-
 475 lean and fuel-rich stoichiometric zone can strongly reduce NO_x content of
 476 exhaust gas. The relative improvement in NO_x emissions with ARC wavy
 477 profile is kept utmost at 13.4-15.9% compared to REF combustor. Thus, micro
 478 combustor design with ARC wavy that provides highly reduced NO_x emission
 479 is determined.

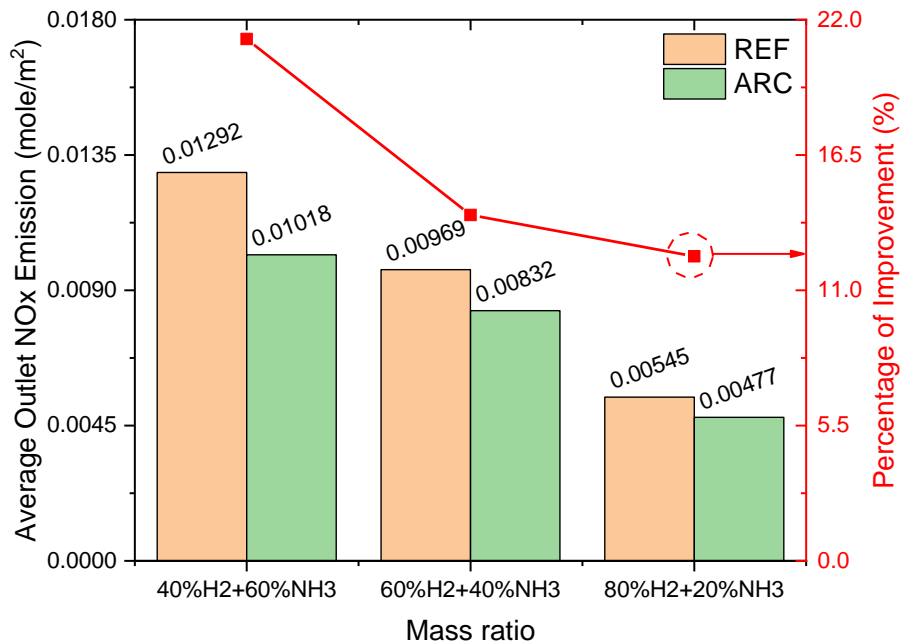


Fig. 12 Variation of NO_x emissions under different mixture ratios of H₂/NH₃.

480 To explore the effect of fuel mixture ratio on NO_x reduction efficiencies, NO_x
481 emission level in different mixture ratios is investigated. Fig. 12 shows the
482 variation of the generated NO_x under different mixture ratios of H₂ and NH₃.
483 All NO_x emission from the outlet of ARC combustor is much lower than that of
484 the REF combustor under the same operating conditions. Average outlet
485 NO_x emission is yielded with an inlet velocity of 12m/s and equivalence ratio
486 of 0.8 for mixture fuel ratio 40%H₂+60%NH₃, 60%H₂+40%NH₃ and
487 40%H₂+60%NH₃ with relative conversion of fuel ammonia to NO_x at 21.2%,
488 14.1% and 12.3% respectively. The relative improvement percentage gradually
489 increases with the increase of the nitrogen fuel mass ratio. The combustor with
490 wavy radiates more heat for energy conversion, so the gas inside is at a lower
491 temperature. This main pathway of NO_x formation mechanism is as following
492 reaction: $N_2 + O \rightleftharpoons NO + N$. Lower temperatures lead to lower O and NO
493 production. It can also be seen from Section 3.4 that the decrease in the
494 proportion of hydrogen will lead to more reduction of NO emission. This is due
495 to the obviously decrease of gas temperature. This also demonstrates that lower
496 gas temperatures are conducive to emissions improvement. From the present
497 simulation data, it is evident that when the combustor improving techniques are
498 advanced such as normal internal wall of REF is substituted with ARC wavy, a
499 subsequent difference in NO_x formation is seen resulting in higher
500 NO_x reduction efficiencies.

4. Conclusions

In this work, in order to enhance the thermal performance and reduce the NO_x emission of the micro-combustor for thermophotovoltaic application, 3D numerical investigations are conducted to evaluate the thermal and NO_x emission performances of the modified micro-combustors with a wavy profile. The model with detailed ammonia/hydrogen/air reaction mechanisms is validated first with experimental data available in the literature. Three different wavy profiles, i.e. TIP, TRI and ARC, are considered, as the NH₃-H₂ mixture flow velocity and equivalence ratio is varied. The impact of such wavy configurations on enhancing the thermal performance, improving flame stabilization and reducing the NO_x emissions is evaluated to obtain an optimum design of a micro-combustor. Key findings are summarized as:

1. The average temperature of the outer wall of the proposed wavy micro combustors (no matter whether the shape is TIP, TRI or ARC) is higher than that of the conventional micro-combustor, regardless of the hydrogen flow velocity. Furthermore, maximum wall temperature is found to be associated with ARC wavy micro combustor. The temperature difference between ARC and the conventional combustor is approximately 36K.
2. The maximum standard deviation of the average temperatures is about 105 K, when the inlet flow velocity is set to 20m/s in reference to the

522 conventional combustor. The minimum non-uniformity coefficient is
523 found to be decreased from 42K to 34K, when considering the ARC
524 wavy configuration. The wavy profile is shown to be more effective on
525 improving temperature uniformity, when the mixture velocity is higher
526 than 12m/s.

527 3. The blowout limits of the conventional micro-combustors operating on
528 the conditions of 60% H_2 +40% NH_3 and 40% H_2 +60% NH_3 are 10 m/s and
529 16m/s, respectively. ARC micro-combustor is found to be more efficient
530 in anchoring flame and expanding the blowout limit to more than 20m/s
531 than the conventional design.

532 4. Regardless of the equivalence ratio, applying the ARC wavy to fuel-lean
533 and fuel-rich stoichiometric zone can strongly reduce NO_x emission
534 found in the exhaust combustion gas. The relative reduction in NO_x
535 emissions with ARC wavy profile is found to be in the range of 13.4-15.9%
536 in comparison with the conventional micro-combustor.

537 5. NO_x emission efficiencies at the combustor outlet under the tested
538 conditions of 40% H_2 +60% NH_3 , 60% H_2 +40% NH_3 and 40% H_2 +60% NH_3
539 are shown to be 21.2%, 14.1% and 12.3% respectively. Further NO_x
540 emission reduction is found to **gradually increase** with the increase of the
541 NH_3 mass ratio.

542 In general, the present work sheds lights on the design of a wavy-shaped micro-
543 combustor with enhanced thermal performance and reduced NO_x emission.

544 **Conflict of interest statement**

545 The authors declared that they have no conflicts of interest to this work.

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