

Experimental Study on Orthogonal Joints in Cross-Laminated Timber with Self-Tapping Screws installed with Mixed Angles

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Abstract

Cross-laminated timber (CLT) is increasingly being used in lateral load resisting systems of multi-storey buildings. Conventional in-plane CLT shear walls can be transformed into CLT core-wall structures with enhanced lateral strength and stiffness when the individual walls are connected orthogonally. In this paper, experimental studies are presented on orthogonal CLT joints with self-tapping screws (STS) installed with mixed angles, i.e. different installation angles between the STS axis and the plane of the CLT surface. A total of 59 orthogonal joint specimens were tested in 9 different configurations to derive the relevant joint performance parameters from monotonic and cyclic tests. The joint specimens used five-layer and seven-layer CLT panels connected by $\emptyset 8\text{mm}$ or $\emptyset 12\text{mm}$ STS. Different ratios of STS installed inclined and STS installed at 90° to the CLT surface were investigated to determine an optimum ratio of STS for enhanced joint performance. It was found that a ratio of one 90° STS for every two inclined STS ensured significant increase in ductility and displacement capacity of approximately three times when compared to specimens with only inclined STS. A minimum moderate ductility was achieved in all test series where the primary failure mode was STS withdrawal. It was found that 90° STS contributed to both strength and stiffness in joints that also contained inclined STS. The average experimental overstrength was 1.7 for most joint configurations. Existing analytical models were adequate in estimating strength but inadequate to estimate stiffness.

Keywords: Cross-laminated timber; orthogonal joints; self-tapping screws; mixed-angle installations; ductility; overstrength

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25 **1 Introduction**

26 **1.1 Cross-Laminated Timber Lateral Load Resisting Systems**

27 Light timber frame (LTF) construction is very popular in North America and Oceania for low- and mid-rise
28 residential buildings up to 6 storeys. In LTF construction, shear walls consist of vertical timber framing
29 members to resist gravity loads and provide stability to the panel-sheathing, which is used as the lateral load
30 resisting system (LLRS) against wind and seismic loads [1]. Post and beam timber frame construction is
31 widely used in Japan for residential buildings up to 3 storeys in which diagonally braced or panel-sheathed
32 shear walls are typically used as LLRS [2,3]. In the last two decades, mass timber construction has been
33 gaining popularity due to the increased availability and cost-efficiency of engineered wood products including
34 cross-laminated timber (CLT) and in part due to the aesthetic appeal and environmental benefits of timber as
35 a construction material [4]. CLT is commonly composed of an odd number of layers of timber boards glued
36 together with a crosswise layup to create large solid timber panels with high in-plane strength and stiffness
37 [5], making it suitable for use as LLRS.

38 CLT buildings commonly use platform construction with in-plane CLT shear walls as the LLRS. In platform
39 construction, each building storey is constructed sequentially and walls are interrupted by horizontal floor
40 elements. CLT wall elements are generally connected together, to CLT floor elements and to the foundation
41 with mechanical connectors and fasteners similar to those standard LTF connectors. The in-plane behaviour
42 of CLT shear walls in platform construction has been well researched in the last two decades [6]. For example,
43 Dujic et al. [7] tested the in-plane behaviour of CLT shear walls; and subsequent research mainly focused on
44 CLT shear walls using standard LTF connectors [8–12]. It is well recognized that CLT panels behave relatively
45 rigid and the joints are critical to govern the shear wall behaviour. Thus, CLT hold-downs and shear
46 connections have also been extensively studied [13,14]. Depending on the vertical joint details between
47 adjacent CLT wall panels, single, coupled, or combined wall behaviour was observed [11].

48 As a consequence of a decade of intense research, the design and construction of CLT structures ‘is no longer
49 a domain for early adopters, but is becoming a part of regular timber engineering practice, also in earthquake-
50 prone regions’ [15]. Globally, updates to buildings codes will allow mass timber constructions up to 8-, 12-,
51 and 18-storeys in Australia, Canada, and the United States, respectively [16–18]. In New Zealand, policies
52 such as the “Zero Carbon” Act [19], and Wood First [20,21] are also promoting the use of mass timber
53 construction. In order to realize taller timber structures, enhanced connection solutions are required that meet
54 increased strength and stiffness demands [22]. However, the load carrying capacity of CLT shear walls is only
55 partially exploited when standard LTF connectors are used, which often have only limited capacity [9].

56 **1.2 Capacity Design and Overstrength**

57 In high seismic areas, taller timber structures will require enhanced connection designs to meet not only
58 increased strength and stiffness requirements, but also to develop adequate energy dissipation as ductile joints

59 in capacity design under seismic loading. In capacity design, the overstrength of ductile joints is required and
 60 often derived experimentally for timber joints with comparison to current analytical strength models [23].
 61 Further, the inherent flexibility of timber structures compared to reinforced concrete leaves a smaller window
 62 to develop this ductility [24]. Current design standards, however, still lack state-of-the-art timber design
 63 technology to realize these taller structures [25].

64 Capacity design requires an understanding of the strength hierarchy among building elements to protect brittle
 65 elements by applying overstrength factors derived from ductile elements along the load path [26]. As CLT
 66 wall elements behave relatively rigid, joints are often designed as ductile elements and then their overstrength
 67 needs to be well understood to protect all non-ductile elements and guarantee system ductility.

68 In timber structures, the discrepancy between analytical design strength and the 95th percentile of the true
 69 strength distribution is generally referred to as overstrength, γ_{Rd} , defined by Jorissen & Fragiaco [27] as:

$$\gamma_{Rd} = \gamma_m \gamma_{an} \gamma_{0.95} = \frac{F_A F_{0.05} F_{0.95}}{F_d F_A F_{0.05}} \quad (1)$$

70 where γ_m is the overstrength attributed to material safety factor; γ_{an} is the overstrength due to conservatism in
 71 analytical models; $\gamma_{0.95}$ is the overstrength due to the experimental joint strength distribution; F_A is the
 72 characteristic strength from analytical models; F_d is the design strength; $F_{0.05}$ is the 5th percentile of strength
 73 distribution; $F_{0.95}$ is the 95th percentile of strength distribution. To date, there have been limited studies to
 74 establish overstrength factors for timber joints [13,28–31]. Timber joints often contain groups of fasteners,
 75 and design codes such as Eurocode 5 [32] introduce an effective number of fasteners to account for a possible
 76 group effect. For ductile joints with dowels, these reductions can lead to conservative strength predictions
 77 making it difficult to quantify overstrength [30,33]. For STS joints, Tomasi et al. [34] reported no group effect
 78 on strength while Hossain et al. [35] provided a conservative recommendation for group effects on both
 79 strength and stiffness. In these studies, the possible non-conservative implication of group effect on
 80 overstrength was not considered.

81 **1.3 Joint Design with Self-Tapping Screws**

82 Joints with self-tapping screws (STS) can offer superior performance when compared to standardized dowel-
 83 type connectors such as nails, bolts or wood screws. STS, manufactured by hardened steel with yield strength
 84 up to 1,000 MPa, are the most popular fastener used in mass timber construction, in part due to their ease of
 85 installation and flexibility in design [5]. STS -while optimized primarily for axial loading- can offer one
 86 reliable solution to meet strength and stiffness demands [36]. Bejtka & Blaß [37] tested STS joints in glued
 87 laminated timber by installing inclined STS and developed an analytical strength model that accounts not only
 88 for the embedding strength of the timber member and the bending capacity of the STS, but also the withdrawal
 89 capacity of the STS and the friction between the members. Their tests showed the increase in strength and
 90 stiffness potential with inclined fully threaded (FT) STS.

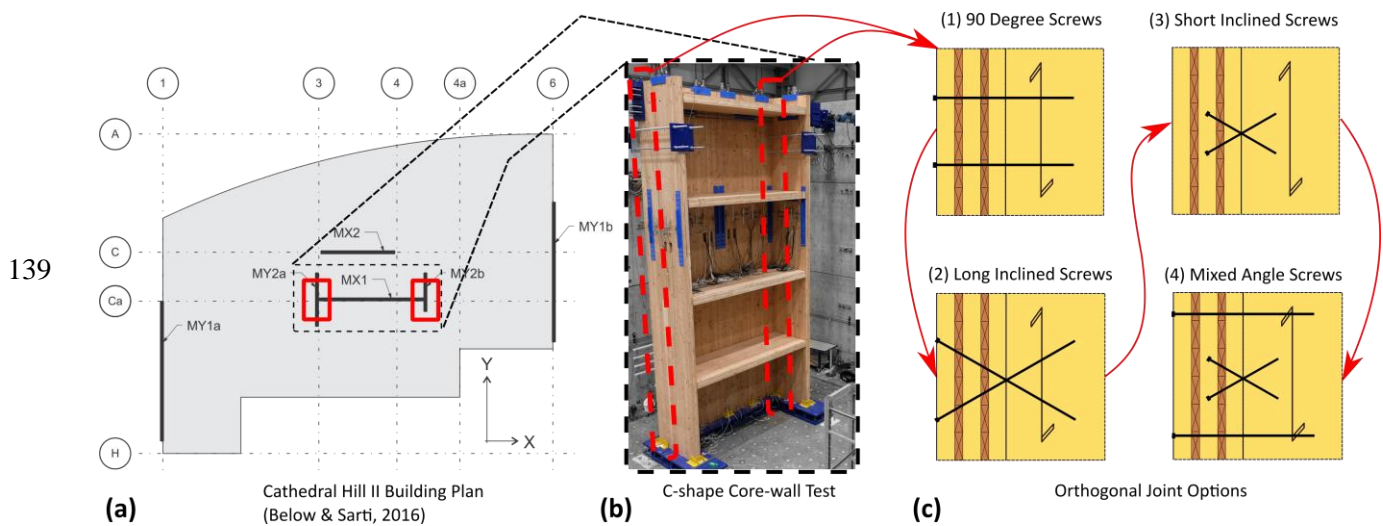
91 The stiffness of inclined STS was studied by Kevarinmaki [38] and a stiffness model was proposed for STS
92 installed at 45° in a shear-tension and cross-wise pattern. Subsequent work by Tomasi et al. [34] extended the
93 existing strength model and developed a stiffness model appropriate for any installation angle. The models
94 were compared against experimental monotonic tests in glued laminated timber. The results showed that the
95 strength model was appropriate and that the stiffness model proposed worked if a “single stiffness” approach
96 was adopted, contrary to the system of springs in series approach proposed by Kevarinmaki [38]. The
97 analytical models used to estimate the strength and stiffness of joints with STS are discussed in more detail in
98 Section 4 of this paper. Tomasi et al. [39] also tested combinations of STS installed inclined and STS installed
99 at 90° to the timber grain, simply called 90° STS, in glued laminated timber and reported promising cyclic
100 performance with mixed installations.

101 The crosswise layup of CLT introduces complexities for joint design. Current design approaches for dowel-
102 type joints in CLT including STS are summarized by Mohammad et al. [40] and Ringhofer et al. [41]. Gavric
103 et al. [29] studied the cyclic performance of 12 different common platform construction STS joints between
104 CLT wall and floor panels. Some tests included in-plane STS spline and lap joints and orthogonal 90° STS
105 joints. It was found that 90° STS joints provided ductile performance in dowel action if recommended spacing
106 and edge distances were followed. The design parameters proposed by Uibel and Blaß [42,43] were
107 appropriate and an overstrength of 1.6 was suggested for the tested STS joints in CLT. Hossain et al. [44]
108 tested butt joints with doubly inclined STS between in-plane CLT panels. The results showed that butt joints,
109 which have a low machining cost, could achieve moderate ductility with a displacement capacity of 8mm
110 under cyclic loading. In-plane CLT lap joints with STS were also studied by considering 90° STS joints,
111 inclined STS joints, and joints with an equal combination of STS in shear and withdrawal [45,46]. The η ratio,
112 i.e. the ratio of STS installed inclined and STS installed at 90° to the timber grain, of 1:1 reported similar
113 findings to Tomasi et al. [39]. With the objective to show that the spatial insertion angle chosen for STS in-
114 plane CLT joints significantly affects the strength, stiffness, and displacement capacity, Loss et al. [47] studied
115 STS in-plane CLT butt joints and compared experimental results to current analytical design models.
116 Satisfactory experimental-analytical agreement was shown for spatially arranged STS strength models but
117 stiffness models were found unsuitable due to the assumptions on individual lateral and axial stiffness
118 components. Increased energy dissipation with increased STS slenderness was also reported [47]. Past
119 research reported that 90° STS act through timber embedment and fastener yielding mechanisms and provide
120 limited stiffness but high displacement capacity, ductility, and energy dissipation. Inclined STS act in
121 withdrawal and provide high strength and stiffness but limited displacement capacity, ductility and energy
122 dissipation. STS joints with η ratio of 1:1 provided promising performance combining high strength, stiffness,
123 ductility, and displacement capacity [39,45–47].

124 **1.4 Cross-Laminated Timber Core-Wall Structures**

125 The feasibility of mass timber core-wall structures has been numerically investigated, either by assuming an
126 orthogonal joint stiffness for a feasibility study [48], or by using small scale inclined STS experimental data

127 as input for orthogonal joint stiffness [49]. There have also been experimental feasibility investigations of
 128 post-tensioned CLT core-walls and the results demonstrated CLT core-walls as a viable LLRS with increased
 129 strength and stiffness [50]. Figure 1 shows the Cathedral Hill II [51] concept design building plan with a
 130 potential core-wall LLRS, to a recently tested 8.6m high C-shaped CLT core-wall at the University of
 131 Canterbury [52], and to four options for orthogonal CLT panel joints with STS. The reported C-shaped CLT
 132 core-wall results indicated that different levels of partial composite action could be achieved based on different
 133 in-plane and orthogonal joint methodologies. STS connections with mixed angle installations for the in-plane
 134 and orthogonal joints offered one effective connection solution with significant increases in strength and
 135 stiffness when compared to an in-plane CLT LLRS system. The complete experimental programme is
 136 introduced Brown et al. [53] which consisted of three post-tensioned CLT shear wall specimens: a single wall,
 137 coupled double wall and C-shaped core-wall. A deep understanding of the orthogonal joints is critical for CLT
 138 core-wall design considering the composite action.



140 *Figure 1: (a) Cathedral Hill II concept design [51] (b) C-shape core-wall test [52] (c) orthogonal joint*
 141 *options with STS*

142 1.5 Objective of research

143 Past experimental work on STS joints in CLT has focussed on common in-plane joints with relatively smaller
 144 fasteners (up to $\phi 10\text{mm} \times 200\text{mm}$) and thin 3- or 5-ply CLT panels. Past orthogonal CLT panel joint tests with
 145 STS were limited to 90° installation angles [29]. As taller timber buildings will require thicker (5-, 7-ply or
 146 greater) CLT panels with larger diameter STS, experimental testing to verify performance is required. While
 147 the use of mixed angle STS with an η of 1:1 has shown promising performance for seismic design [39,45,46],
 148 little work has quantified the impact of inclined to 90° STS η ratio on the joint performance.

149 The primary objective of this study is to evaluate the performance of orthogonal CLT panel joints with varying
 150 mixed angle STS combination ratios, η . These joints are of particular interest for their potential to develop
 151 composite action between orthogonal CLT wall panels, which could transform conventional in-plane CLT

152 LLRS to a core-wall structure with enhanced lateral strength and stiffness. In this study, a total of 59 CLT
153 orthogonal joint tests were performed in 9 different configurations with varying STS η ratio under monotonic
154 and cyclic loading. The different joint configurations were chosen to evaluate which mixed angle STS joint
155 combination could provide enhanced seismic performance and efficiency. The secondary objectives are to
156 compare current analytical strength and stiffness models with the experimental results and to evaluate
157 overstrength. Input parameters for the analytical models are based on both current STS design documents and
158 experimental data from baseline STS withdrawal and lateral load tests.

159 **2 Experimental programme**

160 **2.1 Specimen Description**

161 The test programme is shown in Table 1. A total of 59 joint specimens were tested in nine series with different
162 connection configurations. In labelling each test series, the number indicates the quantity of screws installed
163 in the joints and S, ST, SC, and X indicate different installations: S = 90° STS; ST = shear-tension STS; SC =
164 shear-compression STS, and X = cross-pattern STS (i.e., a combination of shear-tension and shear-
165 compression STS), respectively. Note the test series 16X-400 label is unique and used 16 STS 400mm long,
166 installed in cross-pattern. Three series (2S, 8ST, 8SC) were tested under monotonic (M) loading only, the
167 other six series were tested under both M and reversed cyclic (C) loading. The number of replicates for
168 monotonic tests was three except for the 2S test series which had five replicates; the number of the replicates
169 for cyclic tests was five.

170 Each test series was designed to verify current design models for mixed angle STS as applicable to orthogonal
171 CLT joints. The test programme allowed to assess the performance of orthogonal CLT panel joints with
172 varying mixed angle STS combination ratios. With series 2S, 8ST, 8SC, and 16X, the applicability of existing
173 analytical models to estimate the load-carrying capacity was assessed. By comparing series 16X-400 and 16X,
174 the influence of STS embedment length on strength, stiffness, and failure mode was investigated. Comparing
175 series 16X and 16X+16S (combination of 90° STS and inclined STS) allowed for verification of the increase
176 in displacement capacity and ductility. Finally, series 12X, 12X+4, and 12X+6 aimed to determine the impact
177 of 90° STS on strength, displacement capacity, ductility, stiffness, and energy dissipation. STS slenderness, λ
178 = L/d_c , where L and d_c are the STS length and core diameter, varied from 40 to 80 by considering STS of
179 different L and d_c .

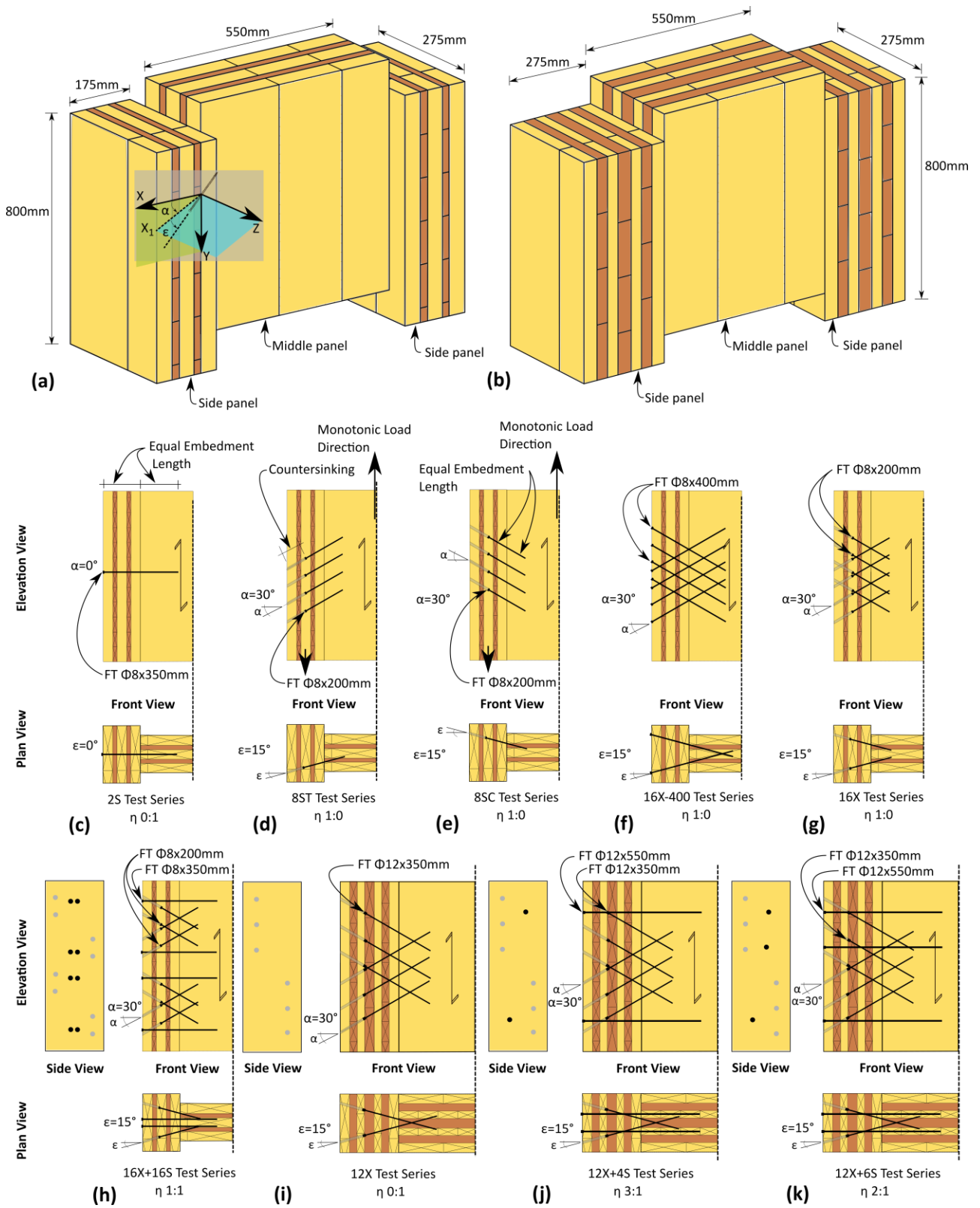
180 The specimens consisted of 5-ply 175mm thick CLT with a layup of 45/20/45/20/45 and 7-ply 275mm thick
181 CLT with a layup of 45/35/35/45/35/35/45, herein simply referred to as CLT5 and CLT7, respectively. The
182 Douglas-fir lamella were graded SG8 with average Modulus of Elasticity of 8 GPa according to NZS3603
183 [54]. After testing, a small piece was removed from each specimen and oven dried to determine density and
184 moisture content. The CLT specimens had an average moisture content of 11%, and the mean and
185 characteristic densities were $\rho_{\text{mean}} = 462 \text{ kg/m}^3$ and $\rho_k = 422 \text{ kg/m}^3$ for CLT5 specimens and $\rho_{\text{mean}} = 457 \text{ kg/m}^3$
186 and $\rho_k = 417 \text{ kg/m}^3$ for CLT7 specimens respectively.

187 SPAX [55] fully threaded (FT) ϕ 8mm STS were used for the CLT5 specimens and FT ϕ 12mm STS were used
188 for the CLT7 specimens. The STS length varied for inclined STS and STS installed at 90° . In series 16X-400
189 the inclined STS length was longer than 90° STS in a similar manner as past research [34,46,56]. However, in
190 all other series to ensure screw withdrawal failure occurred the inclined STS were shorter than 90° STS. The
191 η ratio, i.e. the ratio between inclined STS and STS installed at 90° , varied from 1:0, 3:1, 2:1, 1:1, and 0:1
192 in the joints. Shear-tension and shear-compression screws were both considered inclined STS. The η ratio of 0:1
193 indicated that only 90° STS were used and the η ratio of 1:0 indicated that only inclined STS were used. The
194 η ratios of 3:1, 2:1 and 1:1 indicated the inclined STS to 90° STS ratio. For example, an η ratio of 3:1 meant
195 that for three inclined STS there was one 90° STS. Changing the η ratio accordingly from 1:0, 3:1, 2:1 to 1:1
196 was defined as decreasing the η ratio, and hence increasing the amount of 90° STS in the joint.

197 *Table 1: Experimental Test Programme*

Series	Load Type	Repl.	CLT	Inclined STS			90° STS			Mixed Angle
				Type	λ	Qty.	Type	λ	Qty.	STS Ratio (η)
2S	M	5	CLT5	-	-	-	ϕ 8x350	70	2	0:1
8ST	M	3	CLT5	ϕ 8x200	40	8	-	-	-	1:0
8SC	M	3	CLT5	ϕ 8x200	40	8	-	-	-	1:0
16X-400	M	3	CLT5	ϕ 8x400	80	16	-	-	-	1:0
	C	5	CLT5	ϕ 8x400	80	16	-	-	-	1:0
16X	M	3	CLT5	ϕ 8x200	40	16	-	-	-	1:0
	C	5	CLT5	ϕ 8x200	40	16	-	-	-	1:0
16X+16S	M	3	CLT5	ϕ 8x200	40	16	ϕ 8x350	70	16	1:1
	C	5	CLT5	ϕ 8x200	40	16	ϕ 8x350	70	16	1:1
12X	M	3	CLT7	ϕ 12x350	47	12	-	-	-	1:0
	C	5	CLT7	ϕ 12x350	47	12	-	-	-	1:0
12X+4S	M	3	CLT7	ϕ 12x350	47	12	ϕ 12x550	74	4	3:1
	C	5	CLT7	ϕ 12x350	47	12	ϕ 12x550	74	4	3:1
12X+6S	M	3	CLT7	ϕ 12x350	47	12	ϕ 12x550	74	6	2:1
	C	5	CLT7	ϕ 12x350	47	12	ϕ 12x550	74	6	2:1

198
199 Figure 2 shows the test specimens and joint details of all test series. Figure 2a and Figure 2b provide isometric
200 views and the dimensions for the CLT5 and CLT7 specimens. The ϕ 8mm and ϕ 12mm STS were installed
201 into ϕ 5mm and ϕ 7mm predrilled holes, respectively, to 70% of the screw length with jigs to ensure correct
202 alignment. Each joint specimen had three CLT panels: two side panels and one middle panel. The two side
203 panels were connected to the middle panel with STS installed in a symmetrical layout such that each specimen
204 had two orthogonal joints. Figure 2c - Figure 2k show one joint and half of a test specimen to provide details
205 for each STS layout including the η ratio. The monotonic loading direction is indicated for the 8ST and 8SC
206 test series to show the shear-tension and shear-compression STS respectively. The fastener spacing followed
207 the product ETA [55]. In all test series except 16X-400, the inclined STS were countersunk into the side panels
208 to ensure equal embedment length of the screw into the side panel and the middle panel.



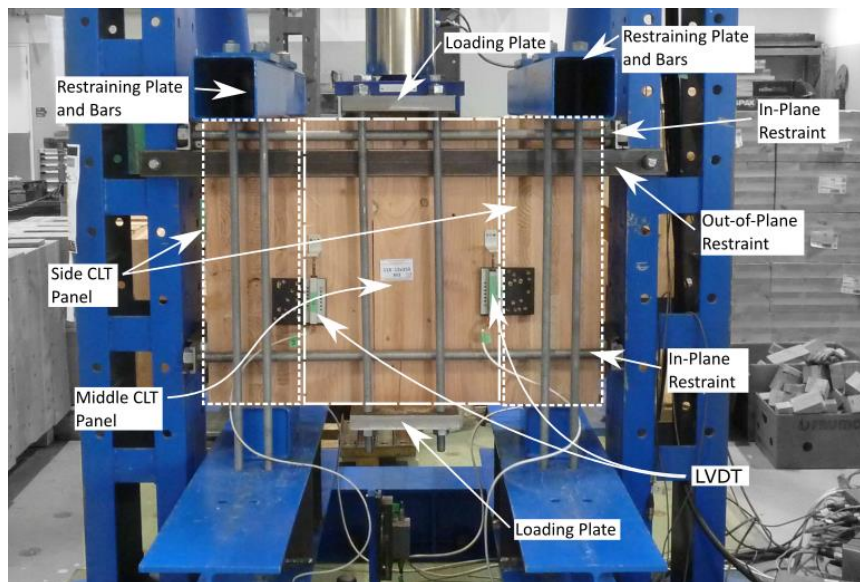
210 Figure 2: (a) CLT5 isometric (b) CLT7 isometric (c) 2S test series (d) 8ST test series (e) 8SC test series (f)
 211 16X-400 test series (g) 16X test series (h) 16X+16S test series (i) 12X test series (j) 12X+4S test series (k)
 212 12X test series

213 In these test series, the inclined STS embedment length was chosen based on single STS withdrawal studies
214 [57] so that the withdrawal strength was sufficiently greater than the STS tensile strength to promote STS
215 withdrawal failure and minimize brittle STS tensile failure. All inclined STS were installed at $\alpha = 30^\circ$ and $\varepsilon =$
216 15° to create a double angle. 90° STS did not have a double angle. For inclined STS, a double angle was
217 implemented for the following reasons: (1) the product ETA [55] requires a minimum angle to the grain of
218 15° for withdrawal capacity; (2) the general embedding strength formulation could be used which is
219 significantly higher than the reduced formulation for STS installed parallel to the CLT plane as per product
220 ETA [55]; (3) significant homogenization is found when STS penetrate more layers [58]; and (4) for an actual
221 core-wall application, the orthogonal joint would be subjected to bi-directional loading and a double angle
222 would provide optimized axial STS loading in either direction.

223 2.2 Methods

224 Figure 3 shows the test setup. A 700 kN capacity hydraulic ram with a load cell was clamped to the middle
225 CLT panel of the joint specimen. The two side CLT panels were fully restrained by steel plates and 4-M20
226 Grade 8.8 threaded rods [59]. Horizontal in-plane movement was also restrained by two sets of steel plates
227 with 4-M36 Grade 8.8 threaded rods placed at the top and bottom of the specimen. Out-of-plane translation
228 and rotation was prevented by a horizontal steel beam with rectangular hollow section that was bolted to the
229 reaction frame.

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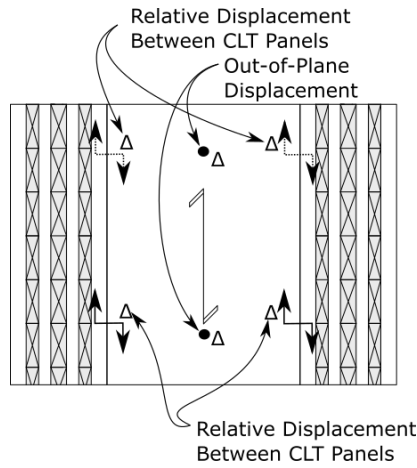


231

Figure 3: Overall test set-up

232 Figure 4 shows the instrumentation used in the testing. Relative displacement between the middle and outer
233 CLT panels was measured with 100mm linear variable displacement transducers (LVDTs) at two points on
234 each shear plane for a total of four measurements. The average joint slip was determined from the four
235 measurements. Out-of-plane displacement was also measured at two points.

236



237

Figure 4: Specimen instrumentation - CLT7 joint specimen shown

238 Test series 2S, 8ST and 8SC were tested under monotonic loading only following EN 26891 [60]. For these
 239 test series if a maximum strength was not reached the joint slip was limited to 15mm following EN 26891
 240 [60]. For the remaining test series, three monotonic tests were performed first to determine the average yield
 241 displacement, Δ_y , that was used as the reference displacement to define the cyclic loading protocol as per EN
 242 12512 [61]. One cycle amplitude at $0.25\Delta_y$ and $0.5\Delta_y$ were performed followed by three cycle amplitudes at
 243 $0.75\Delta_y$, $1.0\Delta_y$, $2.0\Delta_y$, $4.0\Delta_y$, and then increasing multiples of $2.0\Delta_y$ ($6.0\Delta_y$, $8.0\Delta_y$, etc.) until failure as defined
 244 by EN 12512 [61] and explained later. The monotonic loading rate was between 3-6mm/min for a total test
 245 time of 10 to 15min as per EN 26891 [60] and the cyclic loading rate was between 12-18mm/min as per EN
 246 12512 [61]. The results were analysed as per EN 12512 [61] to determine the yield strength F_y , maximum
 247 strength F_{max} , and ultimate strength F_u , the corresponding yield displacement Δ_y , displacement at maximum
 248 strength $\Delta_{F_{max}}$, ultimate displacement Δ_{F_u} , and the elastic stiffness, k . The elastic stiffness was calculated for
 249 the range of the load-slip curve between 10% and 40% F_{max} as per EN 26891 [60]. Herein, the displacement
 250 capacity is synonymous to the ultimate displacement defined as the displacement at which F_u occurred, which
 251 is the post-peak load at 80% of F_{max} . While EN 12512 [61] assesses the ultimate strength F_u to a maximum
 252 slip of 30mm, in 16X+16S, 12X+4S and 12X+6S test series slips greater than 30mm were recorded and they
 253 are presented to illustrate the impact of the η ratio on joint performance. Energy dissipation properties were
 254 derived in terms of equivalent viscous damping following EN 12512 [61]. Ductility, μ , is reported as it is often
 255 defined as a ratio of Δ_{F_u} to Δ_y , as shown in Eq. 2 [27].

$$\mu = \frac{\Delta_{F_u}}{\Delta_y} \quad (2)$$

256 Following the recommendations by Smith et al. [62], the joint was defined as low ductility (LD) for $\mu < 4$, as
 257 moderate ductility (MD) for $4 \leq \mu \leq 6$, and as Ductile (D) for $\mu > 6$.

258 3 Experimental Test Results and Discussion

259 3.1 Overview

260 Table 2 provides a summary of joint performance parameters as mean, X_m , with coefficient of variation (CV),
 261 for each test series. Similar to the results reported by Tomasi et al. [34], ultimate loads were not observed in
 262 both 2S and 8SC test series groups even at large joint slips. Thus, as per EN 26891 [60] the ultimate slip was
 263 limited to 15mm. For the 16X+16, 12X+4, and 12X+6 monotonic test series, F_{max} is reported as the average
 264 load at the first peak on the load-slip curve. In the mixed angle test series under monotonic loading, the load
 265 kept increasing after an initial drop at the first peak and even surpassed the first peak load. The load at the first
 266 peak is required in Section 4 for comparison to analytical models and to derive cyclic overstrength.

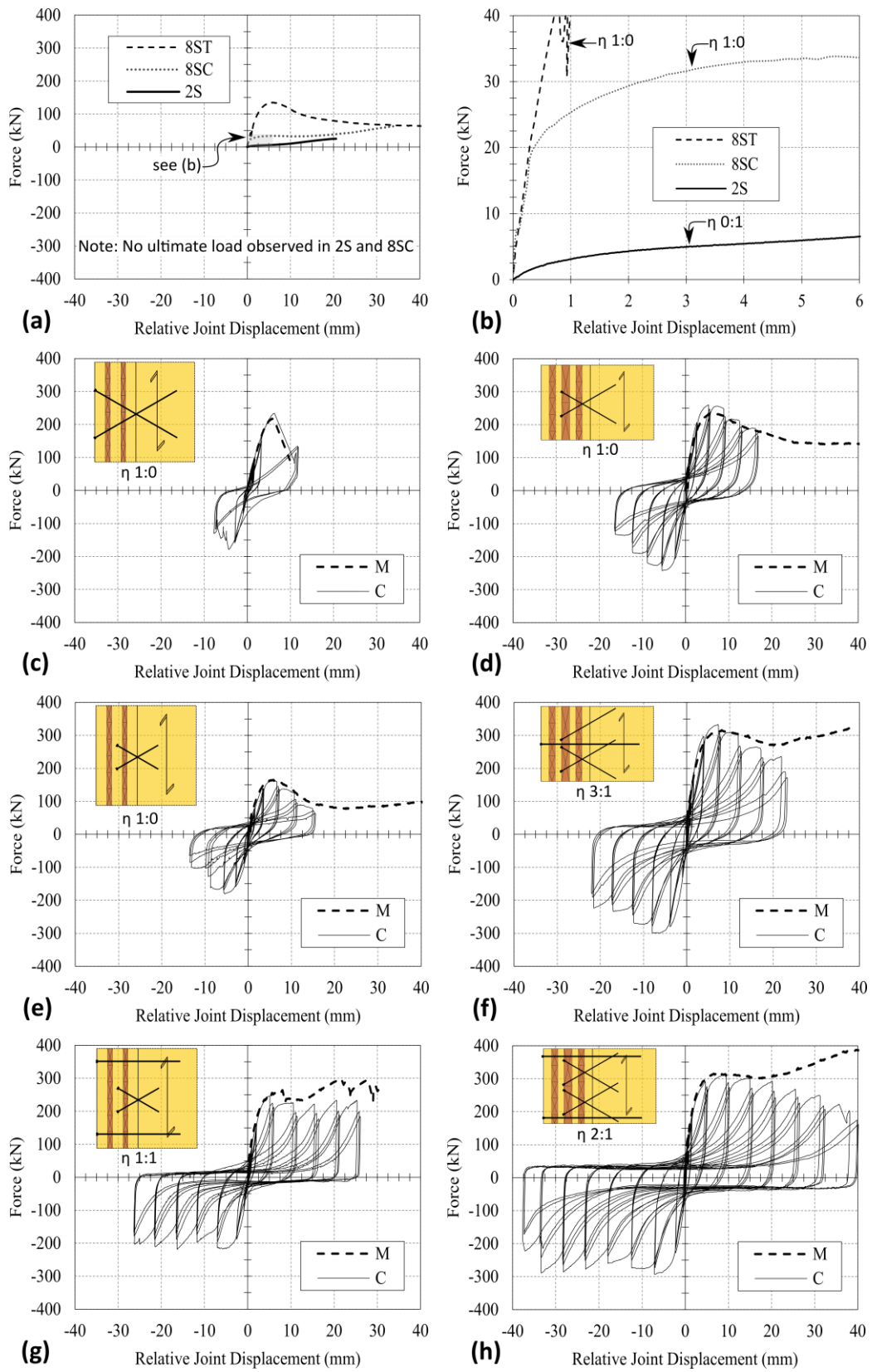
267 *Table 2: Test summary of joint performance factors*

Series	F _y		F _{max}		F _u		Δy		Δ _{max}		Δ _u		K		μ		
	X _m	CV	X _m	CV	X _m	CV	X _m	CV	X _m	CV	X _m	CV	X _m	CV	X _m	CV	
	kN	%	kN	%	kN	%	mm	%	mm	%	mm	%	kN/mm	%	-	%	
2S	M	7	12	7 ¹	-	18	15	6.0	20	6 ¹	-	15 ²	-	0.8	14	- ²	-
8ST	M	121	3	138	2	110	1	2.7	30	5.8	10	11.0	16	45	24	4.2	13
8SC	M	24	13	34 ¹	-	34	6	0.7	22	6 ¹	-	15 ²	-	33	23	- ²	-
16X-400	M	191	2	208	4	167	4	3.8	23	5.9	10	7.1	19	49	19	1.9	15
	C	177	9	202	4	169	7	2.6	15	5.0	7	5.6	11	69	16	2.3	30
16X	M	120	30	153	20	122	20	1.7	35	6.3	29	11.1	16	69	26	7.3	52
	C	134	8	165	5	132	5	1.6	27	5.0	9	7.3	12	83	21	4.9	18
16X+16S	M	190	2	244 ¹	4	251	4	1.9	12	6 ¹	4	26.9	14	92	18	14.4	10
	C	179	6	238	3	190	3	1.7	14	11.0	65	21.8	27	107	15	14.4	24
12X	M	188	4	219	6	176	6	2.4	36	5.8	18	16	18	75	27	7	16
	C	186	11	243	7	195	7	1.6	13	5.7	12	11.6	21	110	8	7.9	33
12X+4S	M	226	11	290 ¹	8	277	20	2	8	8 ¹	7	50.2	32	103	3	25	37
	C	236	7	309	5	247	5	1.7	7	6.6	11	13	9	126	5	7.8	16
12X+6S	M	246	8	314 ¹	6	308	6	2.2	32	8 ¹	4	49.3	4	102	25	23.5	29
	C	215	6	314	5	251	5	1.3	18	9.4	13	24.1	29	151	12	20.2	41

268 Notes: 1 indicates F_{max} chosen as load at first peak for analytical comparison

269 2 as per EN 26891, Δ_u = 15mm and μ not stated as a maximum load was not reached

270 Figure 5 shows the experimental monotonic and hysteresis curves for all test specimens which included two
 271 joints. The force represented the total applied load and the relative joint displacement, slip, was derived by
 272 averaging the data measured from four LVDTs. It was found that the curves of the replicates in each series
 273 were consistent. Therefore, for each test series, one representative monotonic load-slip and one representative
 274 cyclic load-slip curve are provided. The monotonic-load slip curves for the inclined STS show that when
 275 tensile screw failure was avoided, the joint had stable post-peak performance that varied between test series
 276 due to the η ratio. With only inclined STS (η of 1:0), limited displacement capacity was observed. However,
 277 the post-peak displacement capacity was significantly increased by adding 90° STS with the η ratio reduced.



279 *Figure 5: Monotonic and cyclic load-slip curves by test series; (a) and (b) 2S, 8ST, 8SC, (c) 16X-400, (d)*
 280 *12X, (e) 16X, (f) 12X+4S, (g) 16X+16S, (h) 12X+6S*

281 The cyclic load-slip curves showed typical pinching behaviour and stable response in all series other than
282 16X-400, when brittle screw tensile failure occurred. With the addition of 90° STS the displacement capacity
283 increased and the pinching behaviour was more pronounced. Other than for series 16X-400, F_{max} was within
284 10% on average between the positive and negative cycles. The displacement capacity was less consistent
285 between positive and negative cycles of each test.

286 **3.2 Failure Modes**

287 Figure 6 shows typical failure modes for series 16X, 16X+16S and 12X after testing by showing the side and
288 middle panel respectively. Figure 6a - Figure 6f show plastic embedment deformation and the length is
289 indicated at each STS location in mm. Figure 6c and Figure 6d show the longest plastic embedment
290 deformation lengths, indicative of the test series large displacement capacity and the most STS tensile failure
291 as well. Significant plastic embedment deformation is shown by the pronounced pinching behaviour in Figure
292 5g and Figure 5h. Figure 6e shows STS yielding that occurred with each STS removed from the joint specimen.
293 Series 16X-400 had brittle tensile failure of the screws on one shear plane which propagated in a zipper like
294 effect. This is shown by the sudden load drop in Figure 5c. These tests were characterised with low ductility
295 in both monotonic and cyclic loading. The reduced embedment length in series 16X compared to 16X-400 led
296 to a more gradual screw withdrawal as the dominating failure mode, as shown in Figure 5e (series 16X) and
297 Figure 5d (series 16X-400). For the remaining test series, the shortened length of the inclined STS ensured
298 gradual STS withdrawal failure mode. Under monotonic loading tensile screw failure was avoided in most
299 instances and the load increased at larger slips due to the significant rope effect as observed by Tomasi et al.
300 [34]. Under cyclic loading, in some instances tensile screw failure occurred at larger slips but a sudden load
301 drop was avoided. In the mixed angle screw test series, a more complex failure mode similar to that reported
302 by Hossain et al. [45] was observed. The inclined screws provided high initial stiffness. Once screw
303 withdrawal started, a small load drop was observed but the 90° screws became more engaged to carry the load.
304 Under cyclic loading with increased slips, STS tensile failure occurred and the load dropped significantly.
305 However, a progressive zipper-like failure as observed in series 16X-400 did not occur with mixed angle
306 screws and the joint continued to sustain the load. The η ratio influenced the shape of the load slip-curve and
307 failure mode. Figure 5f with η of 3:1 had similar behaviour to the test series in Figure 5d and Figure 5e with
308 η of 1:0. However, Figure 5h with η of 2:1 had similar behaviour to the test series in Figure 5g with η of 1:1,
309 which had been studied for in-plane CLT joints [45,46]. With both decreasing η and increased quantity of
310 screws in the joint, the tensile failure of an individual screw had a lesser effect on the overall joint behaviour.

314 **3.3 Strength**

315 As shown in Table 2, inclined STS joints had significantly higher maximum strength, F_{max} , than 90° STS joints
316 given the specific parameters tested. On a per screw basis and neglecting a possible group effect, F_{max} was
317 approximately five times higher in 8ST test series than 2S test series when considered at the 15mm slip limit.
318 This agrees with past reported research which indicated inclined STS can provide increased strength [34,45–
319 47]. F_{max} in test series 16X was less than 16X-400 due to shorter embedment length. The load-carrying capacity
320 was also less than the superposition of 8ST and 8SC, which indicated that assuming the friction term balanced
321 and was zero in a cross-wise configuration as per Bejtka & Blaß [37] was appropriate in this instance. A
322 progressive increase in F_{max} was observed from series 12X, to 12X+4S, and then to 12X+6S, indicating that
323 the 90° screws contributed to the strength. In all test series except 16X, the CV was notably small (< 8%) and
324 decreased with decreased η and increased screw quantity. The higher CV for F_y (around 9%) when compared
325 to the CV for F_{max} (around 5%) can be attributed to the sensitivity of the method to analyse the load-slip curve
326 [63]. That the CV decreased with decreased η and increased screw quantity indicated the effectiveness of using
327 mixed angle screw combinations, and the importance of testing large multi-fastener joints (up to 16 screws
328 per joint and 32 screws per specimen) to represent actual applications. On average, the ratio of cyclic F_{max} to
329 monotonic F_{max} was 1.04 which was contrary to previous findings by Hossain et al. [45].

330 **3.4 Displacement Capacity and Ductility**

331 The displacement capacity, synonymous to the ultimate displacement or 15mm limit for monotonic specimens
332 which did not reach maximum load, and ductility increased significantly with a maximum η ratio of 2:1.
333 Firstly, the displacement capacity of series 16X+16S with $\eta=1:1$ was three times larger than for series 16X,
334 which confirmed previous findings for in-plane mixed angle STS CLT joints [45,46]. It was found that a
335 minimum number of 90° STS, herein half the number of inclined STS, were required to provide significant
336 influence on joint behaviour. Cyclic ductility was unchanged between series 12X and 12X+4S which indicated
337 that the η ratio of 3:1 was too large and the influence of 90° screws was not significant. However, the series
338 with η of 2:1 (12X+6S) and 1:1 (16X+16S) had high displacement capacity greater than 20mm and cyclic
339 ductility greater than 14, respectively, demonstrating that 90° screws significantly contributed for such ratios.
340 This indicated that a maximum η of 2:1 could be recommended to achieve enhanced joint behaviour. It should
341 be noted that if gradual screw withdrawal of inclined STS was the governing failure mode, moderate ductility
342 was achieved, in agreement with past studies [44,47]. Though, with an η of 1:0 the cyclic displacement
343 capacity is limited and less than 12mm. STS slenderness, $\lambda = L/d_c$, impacted displacement capacity and
344 ductility. When comparing series 12X to 16X, cyclic displacement capacity and ductility increased by a factor
345 of 1.5 with increased λ to 47 from 40 even with larger diameter and fewer STS in the 12X series. Past work
346 by Loss et al. [47] and Sullivan et al. [46] also had increased ductility with increased λ . STS slenderness as an
347 influencing parameter on STS joint performance will be discussed further and the results indicated that
348 slenderness may be a more representative parameter irrespective of STS diameter which had been reported in
349 past work [46]. In all test series, the displacement capacity and ductility were lower under cyclic loading

350 compared to monotonic loading. Displacement capacity and ductility were 35% and 15% lower on average
351 respectively. The yield displacement was also 20% smaller on average under cyclic loading, but was minimally
352 affected by changing η .

353 **3.5 Stiffness**

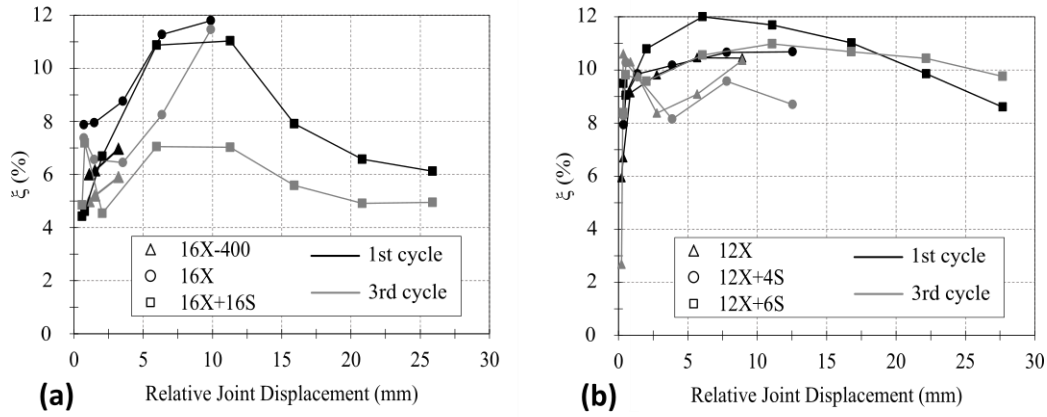
354 The stiffness of inclined STS in ST, SC and X configuration test series was significantly higher than the 90°
355 screws as expected. For series 2S, upon evaluating the elastic portion of the curve in a similar manner to Gavric
356 et al. [29], the derived elastic stiffness was 1.8kN/mm/screw and almost 5 times less than series 16X. While
357 this comparison neglects a possible group effect, it agrees with past reported research which indicated inclined
358 STS provide increased stiffness [34,45–47]. It should be noted that as per EN 12512 [61], the stiffness for 2S
359 was only 0.4kN/mm/screw, but this was significantly influenced by the shape of the load-slip curve and
360 deemed not representative for comparative purposes in this instance. The joint stiffness also increased with
361 decreasing η which indicated that 90° screws impact stiffness. For instance, the progressive increase in cyclic
362 stiffness from series 12X, to 12X+4S, and then to 12X+6S was 110, 126, and 151 kN/mm. Increased STS
363 slenderness, λ , appeared to influence and decrease joint stiffness. The stiffness of series 16X ($\lambda=40$) was 1.2
364 times higher than that of series 16X-400 ($\lambda=80$) which is contrary to the values calculated by the product
365 approval [55], as the STS embedment length in 16X was approximately half that in 16X-400. In the product
366 approval [55] stiffness increases linearly with embedment length. A comparison between series 8ST and 8SC
367 with 16X indicated that there was a contribution from friction to stiffness for shear-tension STS in agreement
368 with past reported work [34,47]. The cyclic stiffness was on average 1.3 times higher than the monotonic
369 stiffness, which could in part be due to the faster cyclic loading rate.

370 **3.6 Energy Dissipation**

371 Energy dissipation was evaluated in terms of equivalent viscous damping, ξ , for the first and third cycle of the
372 load-slip curve at each displacement amplitude. The results are presented as the averages of the replicates of
373 the test series. Figure 7 reports ξ for each displacement amplitude cycle up to the limit of post-peak load at
374 80% of F_{max} in a similar manner to Loss et al. [47]. The results indicated that ξ was directly linked to the
375 associated failure mode. In all test series, the initial increase in ξ at early displacement cycles is indicative of
376 ST and SC screws loaded in withdrawal which have low initial energy dissipation capacity due to high elastic
377 stiffness [44]. Steel tensile failure in 16X-400 resulted in the lowest ξ as expected. For the remaining test
378 series, ξ reached its peak in the two or four times yield displacement amplitude cycles. The increased ξ was
379 due to gradual withdrawal of STS, timber embedment deformation, and STS bending yielding deformation.
380 With decreased η , at large displacement ξ gradually decreased and the difference between ξ_{1st} and ξ_{3rd} was
381 more significant which is typical in dowelled joints or STS installed at 90° with pinched hysteresis loops [29].
382 Previous testing by Loss et al. [47] reported increased ξ to 8% with increased λ from 23 to 30 and noted this
383 positive correlation. In this instance, the average ξ at maximum load was 10% with λ of 40 and 47, which is
384 1.25 times higher than ξ reported by Loss et al. [47] with lower λ . The ξ was found to be similar to values

385 reported by Tomasi et al. [39] with a mixed angle STS installation joint and λ of 41. While increased λ may
 386 increase energy dissipation capacity, λ of inclined screws should be limited to avoid STS tensile failure.

387



388 *Figure 7: Equivalent Viscous Damping of each test series: (a) CLT5 specimens (b) CLT7 specimens*

389 4 Analytical Models and Comparisons with Experimental Results

390 4.1 Experimental Considerations for Models

391 For all inclined STS test series, the angle $\varphi_{\parallel} = \varphi$ between the screw axis and the grain of the longitudinal CLT
 392 layer is related to the screw installation angles $\alpha = 30^\circ$ and $\varepsilon = 15^\circ$, calculated by Eq. 3.

$$\cos \varphi = \cos \varepsilon \sin \alpha \quad (3)$$

393 The angle between the screw axis and the grain of the longitudinal and cross CLT layer is φ_{\parallel} and φ_{\perp}
 394 respectively. The angle between the embedment force and the grain of the longitudinal and cross layer is θ_{\parallel}
 395 and θ_{\perp} respectively. As such, for design purpose the design angles for the longitudinal and cross layer were φ_{\parallel}
 396 $= \varphi = 61^\circ$, $\varphi_{\perp} = 33^\circ$, and $\theta_{\parallel} = \theta = 29^\circ$ for all test series, as shown in Figure 8. Table 3 provides a summary
 397 of key STS properties required for analytical models.

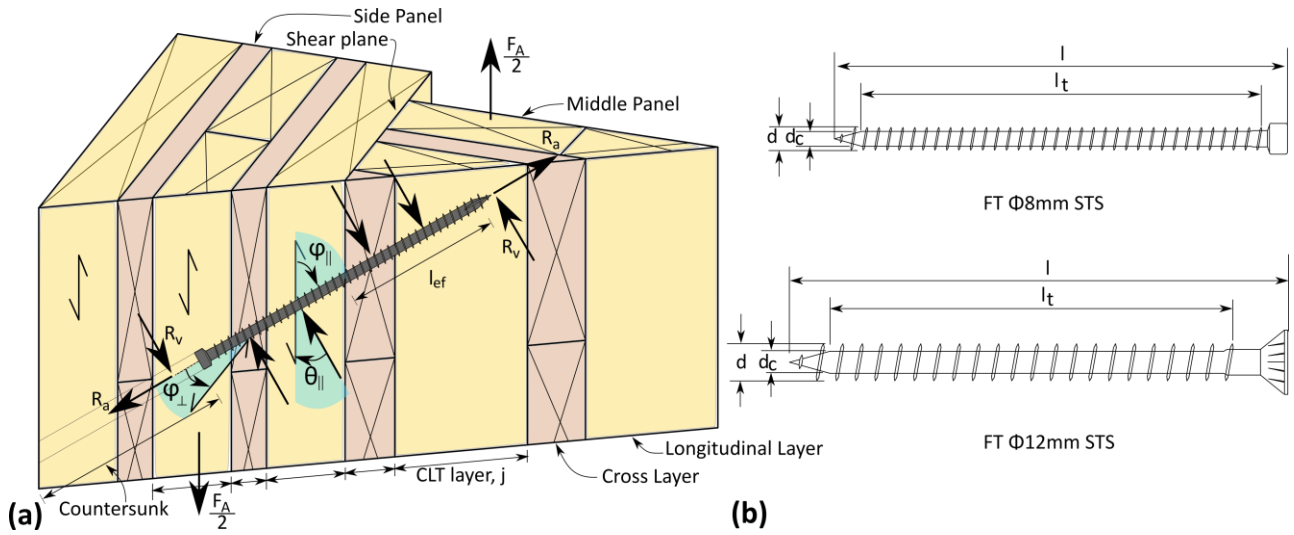
398 *Table 3: Test series STS details*

Series	STS Name	d mm	l mm	d _c mm	l _t (min) mm
16X-400	Ø8x400	8	400	5	375
2S, 16X+16S	Ø8x350	8	350	5	325
8ST, 8SC, 16X, 16X+16S	Ø8x200	8	200	5	185
12X+4S, 12X+6S	Ø12x550	12	550	7.4	525
12X, 12X+4S, 12X+6S	Ø12x350	12	350	7.4	325

399

400

401



402

Figure 8: (a) Isometric of shear-tension $\varnothing 8\text{mm}$ STS in CLT5 (b) key parameters for STS

403

4.2 Strength Model

404

The analytical strength model developed by Bejtka and Blaß [37] with extensions by Jockwer et al. [64] was adapted herein for orthogonal joint design. As STS joint design is not covered by many design standards including New Zealand Timber Structures Standard NZS3603 [54], design guidance within the SPAX ETA [55] was used to determine the withdrawal strength, fastener bending yield moment, and embedment strength component properties for the analytical model because there are differences between different STS ETAs [36]. While current design codes guide designers to a ductile joint by introducing certain factors such as an effective number of fasteners, n_{ef} , Dorn et al. [33] reported this can lead to conservative strength predictions for ductile joints. This makes it hard to quantify the overstrength due to conservatism in analytical models, γ_{an} , and was therefore not considered herein. Accordingly, the withdrawal strength parameter, f_i , is 12.0 and 11.0MPa for the $\varnothing 8\text{mm}$ and $\varnothing 12\text{mm}$ STS respectively, and the bending yield moment, M_y , and embedment strength, $f_{h,\varphi}$ are calculated by Eq. 4 and Eq. 5.

414

$$M_y = 0.15(600)d^{2.6} \quad (4)$$

415

where d is the outer thread diameter. The embedment strength for inclined screws, $f_{h,\varphi}$, is determined as:

$$f_{h,\varphi} = \frac{0.082\rho_k(1 - 0.01d)}{(2.5 \cos^2\varphi + \sin^2\varphi)(k_{90}\sin^2\theta + \cos^2\theta)} \quad (5)$$

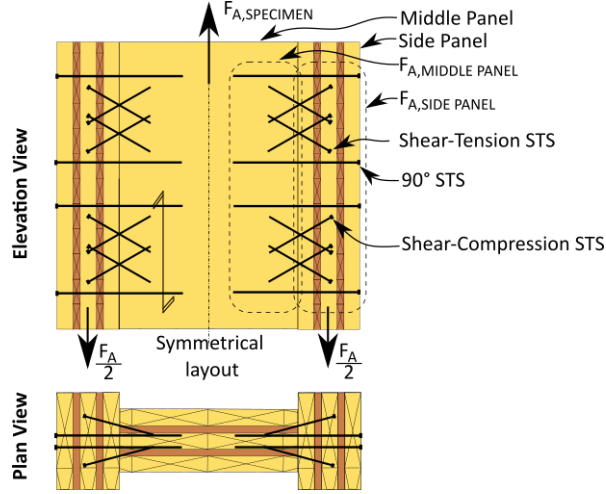
416

where ρ_k is the characteristic CLT density, $k_{90} = 1.35 - 0.015d$, and $\varphi = \varphi_{||}$ and $\theta = \theta_{||}$ as defined above as the angles to longitudinal CLT layer grain direction. For 90° STS in all test series, the embedment strength was $f_{h,S} = 20d^{0.5}$ in the middle panel as the STS was installed parallel to the plane of CLT. With reference to Figure 9, for the orthogonal joint specimen, the side and middle panel, $i = sp$ or mp , strength is determined separately and the minimum governs the specimen capacity as per Eq. 6.

420

$$F_{A,specimen} = 2 * \min \left\{ \begin{array}{l} F_{A,SIDE\ PLATE} \\ F_{A,MIDDLE\ PLATE} \end{array} \right. \quad (6)$$

421 where the factor 2 accounts for both sides of the symmetrical specimen to determine the overall specimen
 422 capacity. In all instances, the side panel strength governed the capacity due to the lower rolling shear strength
 423 of the 45mm layer required for Eq. 12, which will be discussed further.



424

425

Figure 9: Strength calculation illustrated by 16X+16S specimen

426 For a 90° STS acting in dowel action as in the 2S series, the Johansen equations of Eurocode 5 [32] with
 427 consideration for the rope effect are considered as:

$$F_{A,S,i} = R_{v,i} + \min \left\{ \begin{array}{l} R_{a,i}/4 \\ R_{v,i} \end{array} \right. \quad (7)$$

428 where $R_{v,i}$ is shear strength in dowel action and $R_{a,i}$ is the axial strength determined in Eq. 9. The slenderness
 429 of the STS ensures plastic hinges will develop such that $R_{v,i}$ is:

$$R_{v,i} = \sqrt{\frac{2\beta}{1+\beta}} \sqrt{2M_y d_{ef} f_{h,\varphi,i}} \quad (8)$$

430 with M_y as defined before, $d_{ef} = 1.1d_c$ where d_c is the screw core diameter and $f_{h,\varphi,i}$ as defined before. $\beta = f_{h,\varphi,mp}/f_{h,\varphi,sp}$,
 431 is the ratio between embedment strengths on the screw middle panel side and screw side panel side.
 432 In the instance of ST and SC STS $\beta = 1$. The determination of $R_{a,i}$ was as per ETA [55] as the minimum of
 433 either the withdrawal strength or the STS tensile strength:

$$R_{a,i} = \min \left(\frac{f_1 d_{ef}}{1.2 \cos^2 \varphi + \sin^2 \varphi} \left(\frac{\rho_k}{350} \right)^{0.8}; 17,000 (\text{Ø}8\text{mm}), 38,000 (\text{Ø}12\text{mm}) \right) \quad (9)$$

434 where f_t , d , and $\varphi = \varphi_{\parallel}$ are as defined before, and l_{ef} is the screw thread length (mm) in each CLT panel side,
 435 which was half the screw thread length ($l/2$). For each side and middle panel the withdrawal strength
 436 determination, $R_{a,i}$, is the sum of each component determined for each CLT layer, j , penetrated considering
 437 both l_{ef} , φ_{\parallel} and φ_{\perp} with reference to Figure 8. For the shear-tension (ST) STS, the strength is determined as:

$$F_{A,ST,i} = R_{a,i} (\psi \sin \varphi + \cos \varphi) + R_{v,i} (\sin \varphi - \psi \cos \varphi) \quad (10)$$

438 where $R_{a,i}$ and $R_{v,i}$ are the screw axial and shear resistance respectively, $\varphi = \varphi_{\parallel}$ and ψ is the coefficient of
 439 friction, taken as 0.25 for wood-wood surfaces as per Eurocode 5 [32]. For a shear-compression (SC) STS,
 440 the strength is determined similar to a ST STS without the contribution due to friction as:

$$F_{A,SC,i} = R_{a,i}^* \cos \varphi + R_{v,i}^* \sin \varphi \quad (11)$$

441 where $R_{a,i}^*$ was calculated similar to $R_{a,i}$ but included the consideration for “edge effect”. Jockwer et al. [64]
 442 observed that an area from the surface of the timber member was affected by splitting / compression failures
 443 such that a zero stress zone exists up until a certain length, x_1 . In this way l_{ef} is reduced by x_1 , which is defined
 444 as the length from the CLT face with zero embedment and withdrawal capacity and determined as:

$$x_1 = f_{h,\varphi} d_{ef} / 2 \tan \varphi f_{rs} \quad (12)$$

445 where f_{rs} is the rolling shear strength of the applicable layer. In this instance, f_{rs} is 2.2, 1.1 or 0.9MPa
 446 considering the 20, 35, and 45mm layers of the CLT specimens respectively as previously reported by Li et
 447 al. [65]. $R_{v,i}^*$ is determined as per Jockwer et al. [64] as:

$$R_{v,i}^* = \sqrt{2M_y d_{ef} f_{h,\varphi} + (f_{h,\varphi} d_{ef} x_1)^2} - f_{h,\varphi} d_{ef} x_1 \quad (13)$$

448 When ST and SC STS are used together in a X configuration, the strength of a cross-pattern (X) pair of screws
 449 is defined similar to Tomasi et al. [34] as:

$$F_{A,X,i} = (R_{a,i} \cos \varphi + R_{v,i} \sin \varphi) + (R_{a,i}^* \cos \varphi + R_{v,i}^* \sin \varphi) \quad (14)$$

450 where $\varphi = \varphi_{\parallel}$ and the contribution from friction from the ST and SC screw are opposite and balance each
 451 other. For a X + S configuration, in a similar manner to Tomasi et al. [39], by superposition the strength is
 452 determined as:

$$F_{A,X+S,i} = n_x F_{A,X,i} + n_s F_{A,S,i} \quad (15)$$

453 where $F_{A,X,i}$ and $F_{A,S,i}$ are defined above and n_x and n_s are the number of X pair and S STS respectively.

454

455 4.3 Stiffness Model

456 The analytical stiffness model developed by Tomasi et al. [34] and shown in Eq. 16 with work by Kevarinmaki
457 [38] was adapted herein for orthogonal joint design.

$$K_{A,STS} = k_{\perp} \sin^2 \varphi + k_{\parallel} \cos^2 \varphi \quad (16)$$

458 where $K_{A,STS} = K_{A,S}$, $K_{A,ST}$, or $K_{A,SC}$ for a 90° , ST, or SC STS respectively, $\varphi = \varphi_{\parallel}$ as defined before and k_{\perp} is
459 the lateral stiffness component and provided in Eurocode 5 [32] as:

$$k_{\perp} = \frac{\rho_m^{1.5} d_{ef}}{23} \quad (17)$$

460 where ρ_m is the mean characteristic density and d_{ef} is defined previously. The axial stiffness component, k_{\parallel} , is
461 determined following the model proposed by Kevarinmaki [38] which considered the axial stiffness of a screw
462 as a function of the thread stiffness on the middle and side panel side of the screw, similar to a system of two
463 springs in series as:

$$k_{\parallel} = \frac{1}{\frac{1}{k_{ax,sp}} + \frac{1}{k_{ax,mp}}} \quad (18)$$

464 where $k_{ax,sp}$ and $k_{ax,mp}$ are the axial slip modulus of the side and middle panel of the joint respectively,
465 determined by SPAX ETA [55] as:

$$k_{ax} = 25d l_{ef} \quad (19)$$

466 where d and l_{ef} are defined previously. For a 90° STS, $\varphi = 90^\circ$ and the axial component in Eq. 16 reduces to
467 0. For a SC STS, l_{ef} is reduced by x_I as defined before. The overall specimen stiffness is determined as per Eq.
468 20.

$$K_A = n_S K_{A,S} + n_{ST} K_{A,ST} + n_{SC} K_{A,SC} \quad (20)$$

469 where n_S , n_{ST} , and n_{SC} are the number of S, ST, and SC STS respectively. k_{\perp} and k_{\parallel} can also be determined
470 from experimental tests as suggested by Blaß et al. [66] instead of empirical component equations. In this
471 way, the overall specimen stiffness could be determined using experimental STS stiffness for k_{\perp} and k_{\parallel} in Eq.
472 16 and then in Eq. 20 to determine $K_{A,EXP}$.

473 4.4 Experimental-Analytical Comparison and Discussion

474 Table 4 summarizes the experimental-analytical comparison results. Using EN 14358 [67] and assuming a
475 log-normal strength distribution, the 5th and 95th percentile strength, $F_{0.05}$ and $F_{0.95}$, were determined from the
476 cyclic test F_y results. The $F_{0.05}$ was compared to the analytical strength F_A , $\gamma_{an} = F_{0.05}/F_A$, and the experimental
477 overstrength, γ_{rd} , was derived. The average monotonic stiffness, k , was compared to both analytical stiffness

478 K_A , which was derived from empirical component equations and $K_{A/EXP}$, which was derived from experimental
 479 component stiffness results.

480 *Table 4: Experimental-Analytical Comparisons Summary*

Series	F _{0.05} kN	F _{0.95} kN	k kN/mm	F _A kN	K _A kN/mm	γ _{an} -	γ _{0.95} -	γ _{Rd} -	k/K _A -	k/K _{A/Exp.} -
2S	-	-	0.8	6	5	-	-	-	0.2	1.0
8ST	-	-	45	68	38	-	-	-	1.2	2.0
8SC	-	-	33	27	25	-	-	-	1.3	1.7
16X-400	140	222	49	150	104	0.9	1.6	-	0.5	-
16X	108	164	69	81	57	1.3	1.5	2.0	1.2	1.7
16X+16S	154	208	92	132	96	1.2	1.4	1.6	1.0	1.3
12X	139	247	75	140	90	1.0	1.8	1.8	0.8	-
12X+4S	196	284	103	170	104	1.2	1.4	1.7	1.0	-
12X+6S	184	250	102	184	110	1.0	1.4	1.4	0.9	-

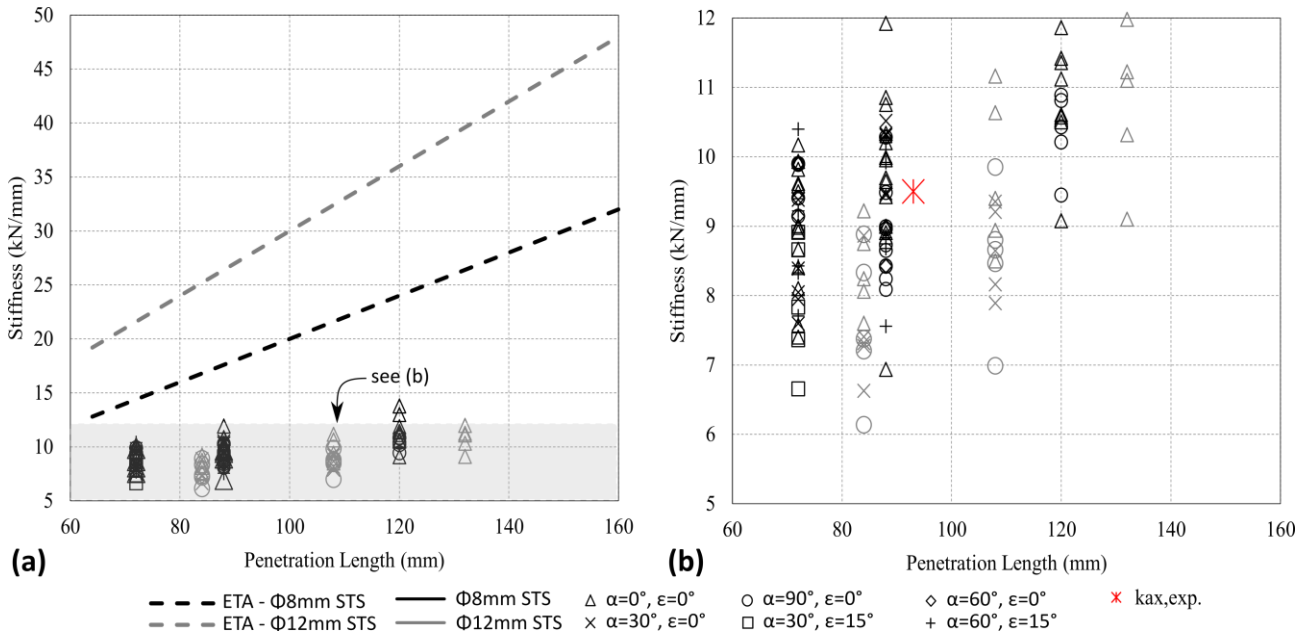
481
 482 The average γ_{an} , which is the ratio between the experimental 5th percentile strength and analytical strength,
 483 was 1.1. This shows that the analytical strength model considered in Section 4.2 and used herein was
 484 acceptable. The appropriateness of using superposition in Eq. 15 to determine for example $F_{A,16X+16S}$ was
 485 calculated considering the 16X and 2S test series average monotonic results. $F_{2S} = 7\text{kN}$ was considered at
 486 6mm joint slip because for series 16X and 16X+16S, peak load occurred at approximately 6mm joint slip. For
 487 comparison, $F_{max,16X-M} + 8F_{2S} = 209\text{kN}$ which is within 15% of $F_{max,16X+16-M}$. Therefore, superposition of
 488 mixed angle screws provided reasonable predictions in this instance, as was reported by Tomasi et al. [39].
 489 However, a strength prediction model which can account for the significantly different stiffness of inclined
 490 STS and 90° STS is needed to give more accurate prediction results.

491 The experimental stiffness, k , was compared to analytical stiffness, as per Eq. 20, considering both empirical
 492 component equations to determine K_A and experimental component test results for k_{\perp} and k_{ax} to determine
 493 $K_{A/EXP}$. In general, the analytical stiffness model was inadequate. Although the k/K_A ratio showed that the
 494 analytical model appeared to be working well, as noted by Loss et al. [47], the model is very sensitive to the
 495 components k_{\perp} and k_{ax} . In determining K_A , k_{\perp} was determined as per Eurocode 5 [32] which had been derived
 496 for a traditional wood screw and does not consider the screw type, insertion angle and length of the STS [47].
 497 Reported results herein of $k_{2S}/K_A = 0.2$, which were similar to past reported lateral stiffness [46,47], indicated
 498 that Eq. 17 from Eurocode 5 [32] is not appropriate for STS. As k_{\perp} contributed to both inclined STS and 90°
 499 STS this affected the analytical stiffness model. Further, it has been reported that the axial slip modulus, k_{ax} ,
 500 equations used can provide significant differences up to 500% depending on the screw diameter and insertion
 501 length (Ringhofer, 2017). For example, the $\emptyset 12 \times 350\text{mm}$ screws could have $k_{ax} = 48.9\text{kN/mm}$ or 9.8kN/mm if
 502 Eq. 19 or if the equation used in Loss et al. [47] of $k_{ax} = 780d^{0.2}l_{ef}^{0.4}$ was used. As recommended by Blaß et
 503 al. [66], as a second comparison, $K_{A/Exp.}$ was determined using experimental STS stiffness for k_{\perp} and k_{ax} and it

504 was compared to experimental results as $k/K_{A/Exp}$. in Table 4. To determine $K_{A/Exp}$, k_{\perp} was 1.8kN/mm as reported
 505 in Section 3.5 and k_{ax} was 9.5 kN/mm as per Figure 10 to determine k_{\parallel} as per Eq. 18. Figure 10 shows the
 506 reported experimental k_{ax} results of 187 STS withdrawal tests with $\varnothing 8$ mm and $\varnothing 12$ mm STS at various angles
 507 to the grain and penetration lengths in comparison to Eq. 19. The average $k/K_{A/Exp}$. ratio of 1.7 indicated that
 508 the analytical stiffness model of Tomasi et al. [34] underestimated the observed experimental stiffness. That
 509 Tomasi et al. [34] observed similar findings with $k/K_A \approx$ up to 2.0 when k_{\parallel} was determined as per Kevarinmaki
 510 [38] suggests that further research is required to capture the joint stiffness of STS installed at varying
 511 inclinations to grain.

512

513



514

Figure 10: Experimental STS withdrawal stiffness with comparison to empirical equation [57]

515

4.5 Overstrength Discussion

516 The overstrength of each test series was calculated as per Eq. 1 assuming $\gamma_m = 1.0$ as per Eurocode 8 [69]. As
 517 per Table 4, the average cyclic experimental γ_{Rd} was 1.7 excluding the 16X-400 test series as brittle STS tensile
 518 failure occurred. This γ_{Rd} was comparable to past experimental overstrength factors for timber joints [13,28–
 519 31]. The slightly higher experimental overstrength reported herein could in part be due to the relatively small
 520 sample size as the average $\gamma_{0.95}$ was 1.5. The average γ_{an} was 1.1 which is comparable to $\gamma_{an} = 1.18$ [27] and
 521 $\gamma_{an} = 1.06$ [30]. In a similar manner to Ottenhaus et al. [30], the analytical component overstrength component
 522 can be determined as per Eq. 21.

$$\gamma_{an} = \gamma_{an,cyc} \gamma_{fh} \gamma_{an,f1} \gamma_{an,My} \quad (21)$$

523

where $\gamma_{an,cyc}$. is the ratio of cyclic loading to monotonic loading, $\gamma_{an,fh}$ is overstrength from embedment strength
 524 formulation, $\gamma_{an,f1}$ is overstrength from the withdrawal strength parameter, and $\gamma_{an,My}$ is the overstrength from

525 the STS yield moment formulation. The average overstrength observed under cyclic loading, $\gamma_{an,cyc.}$, was 0.98.
526 For instance, the 16X test series $\gamma_{an,fl}$ was 1.5 when considering the experimental withdrawal strength
527 parameter reported by Brown et al. [57]. A parametric component study of embedment strength and yield
528 moment determination could also define $\gamma_{an,fb}$ and $\gamma_{an,My}$ respectively. It is important to note that the
529 experimentally determined overstrength of this study should only be used for this particular tested joint. A
530 generic analytical component strength overstrength approach such as that developed by Ottenhaus et al. [30]
531 was beyond the scope of this study, though it could provide a strong alternative to costly experimental testing.

532 **5 Conclusions**

533 The paper investigated the performance of orthogonal joints between CLT panels with varying mixed angle
534 STS combination ratios, η , for the purpose of developing enhanced joints between CLT wall panels. A total
535 of 59 specimens consisting of two CLT layups and different STS sizes were tested under monotonic and cyclic
536 loading to determine strength, displacement capacity, ductility, stiffness and overstrength and to compare to
537 analytical predictions. The key findings are summarized as follows:

- 538 • Mixed angle STS joints had increased joint displacement capacity, ductility and energy dissipation when
539 compared to inclined only STS joints.
- 540 • Based on the test results, a maximum inclined STS to 90° STS η ratio of 2:1 led to more efficient design
541 than the η ratio of 1:1. The 2:1 η ratio ensured high displacement capacity exceeding 20mm, whereas a
542 larger 3:1 or 1:0 η ratio had displacement capacity limited to 13mm or less respectively. Displacement
543 capacity is critical to develop ductility and hysteretic damping under seismic loading. The 2:1 η ratio
544 ensured rope effect by 90° STS at increased joint displacement was significant enough to maintain post-
545 peak strength above 80% F_{max} .
- 546 • Strength and stiffness of the mixed angle STS joints was affected by 90° STS. For example, in cyclic series
547 12X, 12X+4S, and 12X+6S, F_{max} was 195, 247, and 251kN and k was 110, 126 and 151kN/mm
548 respectively. Peak strength in inclined only and mixed angle test series occurred at similar displacements,
549 which can provide one reason for using superposition to estimate the joint strength.
- 550 • The average experimental overstrength, γ_{Rd} , was 1.7 excluding 16X-400 where brittle tensile failure
551 occurred. Existing analytical models were found to be adequate in estimating the joint strength using
552 superposition to determine the strength of joints with STS of mixed angles with different stiffness. Further
553 work should verify the suitability of such method.
- 554 • Analytical models for estimating joint stiffness were found to be inadequate, especially when considering
555 experimental results for k_{\perp} and k_{ax} from 90° STS and single STS withdrawal tests.
- 556 • The preferred failure mode for inclined STS joints is gradual screw withdrawal. It is critical to limit the
557 STS thread embedment such that progressive zipper-like tensile failure is avoided. Except for series 16X-
558 400, screw withdrawal failure was the dominant failure mode which led to moderate to high ductility, $\mu \geq 4$,

559 even in the test series with only inclined STS. However, the displacement capacity is limited in joints with
560 inclined only STS and significantly less than joints with η ratio of 0:1, 1:1 and 2:1.

- 561 • STS slenderness ratio, λ , was found to influence the joint displacement capacity, ductility, stiffness and
562 energy dissipation capacity. $\varnothing 12$ mm STS with higher λ had increased displacement capacity, ductility, and
563 energy dissipation capacity in terms of equivalent viscous damping than $\varnothing 8$ mm STS. This highlighted the
564 importance of testing larger diameter STS with various λ in larger 7-ply (275mm thick) CLT panels which
565 may be required in taller timber buildings.

566 While the results presented herein showed mixed angle STS installations could provide enhanced displacement
567 capacity, ductility and energy dissipation, further work is needed to optimize design. Inclined STS without
568 countersinking but with controlled thread embedment length on the STS tip side member could be investigated
569 for similar enhanced performance. A possible group effect for ductile STS joint design should be further
570 investigated. This study provided fundamental information for a better understanding of mixed angle STS
571 joints in orthogonal CLT panels such that in-plane CLT LLRS could transform to core-wall structures with
572 enhanced strength, stiffness and energy dissipation capacity.

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